	EASA		PROPOSAL TO ISSUE AN
		PAD No.: 15-043R1	
	***	with Regulation (EC) No 216/2008 on	rective (PAD) is issued by EASA, acting in accordance behalf of the European Community, its Member States at participate in the activities of EASA under Article 66 of
	In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address spectron, prior to the consultation closing date indicated.		ed below.
	Design Approval		Type/Model designation(s):
	AIRBUS HELICO	PTERS DEUTSCHLAND GmbH	MBB-BK117 C-2 helicopters
	TCDS Number:	EASA.R.010	
	Foreign AD:	Not applicable	
	Supersedure: None		
	ATA 88	Wiring / Harness – Collective Lever Wiring Harness – Inspection	
	Manufacturer(s):	Airbus Helicopters Deutschland G Deutschland GmbH), Airbus Helic LLC).	mbH (AHD) (formerly Eurocopter opters Inc. (formerly American Eurocopter
	Applicability:	MBB-BK117 C-2 helicopters, all s	erial numbers (s/n) up to s/n 9708 inclusive.
	Reason:	be fully engaged by the pilot in the	n an MBB-BK117 C-2 helicopter could not e flight position. Further investigation found a ne collective lever wiring harness, caused by nrinkable sleeve.
		This condition, if not detected and harness and consequent malfunct	corrected, could result in chafing of the ioning of affected systems.
			ondition, AHD issued Alert Service Bulletin to provide instructions for inspection and
		heat-shrinkable sleeve installed o	his AD requires a one-time inspection of the n the collective lever (pilot) wiring harness and, the accomplishment of the applicable
		revised following operator feedbac	HD ASB MBB-BK117 C-2-88A-010 was ck to also require inspection of the transparent the collective lever de-installed. This PAD is Revision 1.

	Effective Date: [TBD: 14 days after final AD issue date]			
-	Required Action(s) and Compliance Time(s):	<ul> <li>Required as indicated, unless accomplished previously:</li> <li>(1) Within the compliance time as specified in Table 1 of this AD, accomplish a one-time inspection of the heat-shrinkable and transparent sleeves installed on the collective lever (pilot) wiring harness in accordance with the instructions of AHD ASB MBB-BK117 C-2-88A-010 Revision 1.</li> <li>Table 1 – One-time inspection</li> </ul>		
		Helicopter configurationCompliance time (after the effective date of this AD)		
		Externally mounted hoist and/or emergency flotation system installed Within 100 FH or three months, whichever occurs first		
		No externally mounted hoist and/or emergency flotation system installed		
		<ul> <li>(2) If, during the inspection as required by paragraph (1) of this AD, any discrepancy is found, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of AHD ASB MBB-BK117 C-2-88A-010 Revision 1.</li> </ul>		
	Ref. Publications:	AHD ASB MBB-BK117 C-2-88A-010 Revision 1 dated 16 April 2015. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.		
	Remarks:       1. This Proposed AD will be closed for consultation on 14 May 2015.         2. Enquiries regarding this PAD should be referred to the Safety Inform Section, Certification Directorate, EASA. E-mail: ADS@easa.europa.         2. Enquiries regarding the technical context of the requiries regarding the technical context of the requirement of the re			
		<ol> <li>For any question concerning the technical content of the requirements in this PAD, please contact: Airbus Helicopters Deutschland GmbH, Industriestrasse 4, 86607 Donauwörth, Federal Republic of Germany Telephone: + 49 (0)151 1422 8976; Facsimile: + 49 (0)906 71 4111.</li> </ol>		

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