Notification of a proposal to issue an Airworthiness Directive

PAD No.: 16-101R1

Issued: 11 August 2016

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) 216/2008 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.
All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name:	Type/Model designation(s):
AIRBUS	A310 aeroplanes

Effective Date:	[TBD - standard: 14 days after AD issue date]
TCDS Number(s):	EASA.A.172
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2014-0124R1 dated 23 June 2015 (the requirements applicable to A310 aeroplanes)

ATA 05 – Time Limits / Maintenance Checks – Airworthiness Limitations Section Part 2 – Amendment

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Manufacturer(s):
Airbus (formerly Airbus Industrie).

Applicability:
Airbus A310 aeroplanes, all certified models, all manufacturer serial numbers.

Reason:
The airworthiness limitations for Airbus A310 aeroplanes, which are approved by EASA, are currently defined and published in the Airbus A310 Airworthiness Limitations Section (ALS) document(s). These instructions have been identified as mandatory actions for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.

EASA previously issued AD 2014-0124 (later revised) to require the actions as specified in Airbus A310 Airworthiness Limitation Item (ALI) Document at issue 08.

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Since EASA AD 2014-0124R1 was issued, Airbus replaced ALI Document issue 08 with A310 ALS Part 2 Revision 01 and then published the A310 ALS Part 2 Variation 1.1 and Variation 1.2, to introduce more restrictive maintenance requirements and/or airworthiness limitations.

For the reason described above, this AD retains part of the requirements of EASA AD 2014-0124R1, which will be cancelled after publication of a Final AD (currently PAD 16-102R1, applicable to A300-600 aeroplanes), and requires accomplishment of the actions specified in Airbus A310 ALS Part 2 Revision 01, ALS Part 2 Variation 1.1 and ALS Part 2 Variation 1.2 (hereafter collectively referred to as 'the ALS' in this AD).

This PAD was revised to introduce the ALS Part 2 Variation 1.2.

Required Action(s) and Compliance Time(s):
Required as indicated, unless accomplished previously:

(1) From the effective date of this AD, within the thresholds and intervals as specified in the ALS (see Note 1 of this AD), accomplish all maintenance tasks, as applicable to the aeroplane configuration, described in the ALS. In case of inconsistencies between a requirement of ALS Part 2 Revision 01 and a requirement of ALS Part 2 Variations 1.1 or 1.2, the requirement of ALS Part 2 Variations must be accomplished.

Note 1: For the purpose of this AD, the thresholds and intervals as defined in the ALS includes specific compliance times, as defined in the Compliance Time pages of the ALS for certain tasks.

(2) **Corrective Action(s)**: In case of finding discrepancies during accomplishment of any task as required by paragraph (1) of this AD, before next flight, accomplish the applicable maintenance procedures for corrective action(s) in accordance with the applicable Airbus maintenance documentation. If a detected discrepancy cannot be corrected by using existing Airbus instructions, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly.

(3) **Aircraft Maintenance Programme (AMP) Revision**: Within 12 months after the effective date of this AD, revise the approved AMP, on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane, by incorporating the tasks and associated thresholds and intervals described in the ALS, as applicable to the aeroplane configuration.

(4) **Credit**: For an AMP that, on the effective date of this AD, is already updated to incorporate the maintenance tasks and life limitations as specified in ALI Document issue 08, reference AI/SE-M2/95A.1309/07, that action ensures (see Note 2 of this AD) the continued accomplishment of those tasks and limitations.
Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations, as applicable to aeroplane model and depending on aeroplane configuration, as defined in, and within the compliance times (see Note 1 of this AD) as specified in the ALS to comply with paragraph (1) of this AD.
For that AMP, it is acceptable to incorporate the new and/or more restrictive tasks and limitations, as defined in the ALS, into the AMP to comply with paragraph (3) of this AD.

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(5) **Recording AD compliance**: When the AMP of an aeroplane has been revised as required by paragraph (3) or (4) of this AD, as applicable, that action ensures (see Note 2 of this AD) continued accomplishment of the tasks as required by paragraphs (1) and (2) of this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.

Note 2: For affected Airbus A310 aeroplanes registered in Europe, complying with the approved AMP, as specified in paragraph (3) or (4) of this AD, as applicable, is required by Commission Regulation (EU) No 1321/2014, Part M.A.301, paragraph 3.

Ref. Publications:
A310 ALS Part 2 Revision 01 dated 07 August 2015.

A310 ALS Part 2 Variation 1.1 dated 25 January 2016.

A310 ALS Part 2 Variation 1.2 dated 22 July 2016.

The use of later approved revisions or variations of these documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 25 August 2016.

2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

3. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS SAS – EIAW (Airworthiness Office)
E-mail: continued.airworthiness-wb.external@airbus.com.

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