## EASA

## **AIRWORTHINESS DIRECTIVE**

AD No.: 2011-0211

Date: 31 October 2011

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Type Approval Holder's Name :		Type/Model designation(s):
AIRBUS		A330 and A340 aeroplanes
TCDS Number :	EASA.A.004, EASA.A.015	
Foreign AD :	Not applicable	
Supersedure :	This AD supersedes EASA	A AD 2008-0223 dated 15 December 2008.
	Landing Cook Mad	A South Cook (MIC) Posis Poses
ATA 32	Inspection / Repair	n Landing Gear (MLG) Bogie Beam –
Manufacturer(s):	Airbus (formerly Airbus	s Industrie)
Applicability:		es, models -201, -202, -203, -223, -223F, -243, -243 , -322, -323, -341, -342 and -343, all manufacturer
		es, models -211, -212, -213, -311, -312, -313, -541, all manufacturer serial numbers.
Reason:	was found deformed ar	aintenance inspection on the MLG, the bogie stop pand cracked. Upon removal of the bogie stop pad for beam was also found cracked.
		on indicates that an overload event has occurred and of the crack was evident.
		crack has subsequently been found on another der a bogie stop pad which only had superficial paint
	departing the runway o	etected and corrected, could result in the aeroplane or to the bogie detaching from the aeroplane or gear dall constitute unsafe conditions at speeds above 30
	inspections under the b	asure, EASA AD 2008-0223 required one-time deta cogie stop pad of both MLG bogie beams and, in ca e is detected, to apply the associated repair.
	Numerous bogie stop p	oad were found corroded and a few cracked as a re

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of the one-time inspection required by EASA AD 2008-0223 on A330. A340-200 and A340-300 aeroplanes. For the reasons described above, this AD, which supersedes EASA AD 2008-0223: retains the initial inspection requirement of EASA AD 2008-0223 for A330, A340-200 and A340-300 aeroplanes. introduces a repetitive detailed visual inspection for A330, A340-200 and A340-300 aeroplanes. retains the requirement of EASA AD 2008-0223 for A340-500 and A340-600 aeroplanes, for which further mandatory requirements might follow in future depending on the results of the one-time mandatory inspection in place. 14 November 2011 Effective Date: Required action(s) Required as indicated, unless accomplished pr and Compliance (1) Within the inspection threshold indicated in Table 1 of this AD, and Time(s): depending on the MLG bogie beam condition at the effective date of this AD, perform detailed inspections of the left-hand (LH) and righthand (RH) bogie beams in the area of the bogie stop pad for detection of deformation of damage and do before further flight the applicable corrective actions Table 1 – MLG Bogie Beam Inspection **MLG Bogie Beam** Inspection Threshold Required condition at the Instructions effective date of this AD Airbus SB A330-32-MLG bogie beams Upon accumulation by the which have not been MLG bogie beam of 2 500 3220 or SB A340inspected in flight cycles (FC) or 22 32-4264 or SB accordance with the months, whichever occurs A340-32-5087, as instructions of Airbus first, since its first flight on applicable to Service Bulletin (SB) an aeroplane or since its aeroplane model A330-32-3220, SB first flight after its last A340-32-4264, or SB overhaul, as applicable, but A340-32-5087, as no later than 40 months applicable to aeroplane since its first flight on an aeroplane or since its first model flight after its last overhaul. MLG bogie beams a) Within 3 months after the Airbus SB A330-32which have been effective date of this AD, 3248 or Airbus SB inspected in A340-32-4286, as accordance with the applicable to instructions of Airbus b) Within 2 500 FC or 24 aeroplane model. SB A330-32-3220, or months, whichever occurs SB A340-32-4264, as first, accumulated by the applicable to aeroplane MLG bogie beam since its model. first flight after the last accomplishment of Airbus SB A330-32-3220 or A340-32-4264, as applicable to aeroplane model, whichever occurs later between a) and b).

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	(2) Thereafter, except for MLG bogie beams which have been inspected in accordance with the instructions of Airbus SB A340-32-5087, repeat the inspection required by paragraph (1) of this AD at intervals not to excee 2 500 FC or 24 months, whichever occurs first, from the last inspection, and do before further flight the applicable corrective actions in accordance with the instructions of Airbus SB A330-32-3248 or Airbus SA340-32-4286, as applicable to aeroplane type.	
	(3) Report to Airbus the results of the inspection, whatever they are, done in accordance with the instructions of Airbus SB A330-32-3220 or Airbus SB A340-32-4264 or Airbus SB A340-32-5087, as applicable to aeroplane model and as required by paragraph (1) of this AD, within 90 days from the accomplishment date of the inspection.	
Ref. Publications:	Airbus Service Bulletin A330-32-3220 at original issue dated 10 October 2008	
	Airbus Service Bulletin A340-32-4264 at original issue dated 10 October 2008	
	Airbus Service Bulletin A340-32-5087 at original issue dated 10 October 2008	
	Airbus Service Bulletin A330-32-3248 at Original issue dated 05 October 2011	
	Airbus Service Bulletin A340-32-4286 at original issue dated 05 October 2011	
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.	
Remarks:	If requested and appropriately substantiated, EASA can approve     Alternative Methods of Compliance for this AD.	
	<ol> <li>The required actions and the tisk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication.</li> </ol>	
	<ol> <li>Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a>.</li> </ol>	
	<ol> <li>For any questions concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EAL Fax: + 33 5 61 93 45 80 or + 33 5 61 93 44 51; E-mailx trworthiness.A330-A340@airbus.com.</li> </ol>	

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