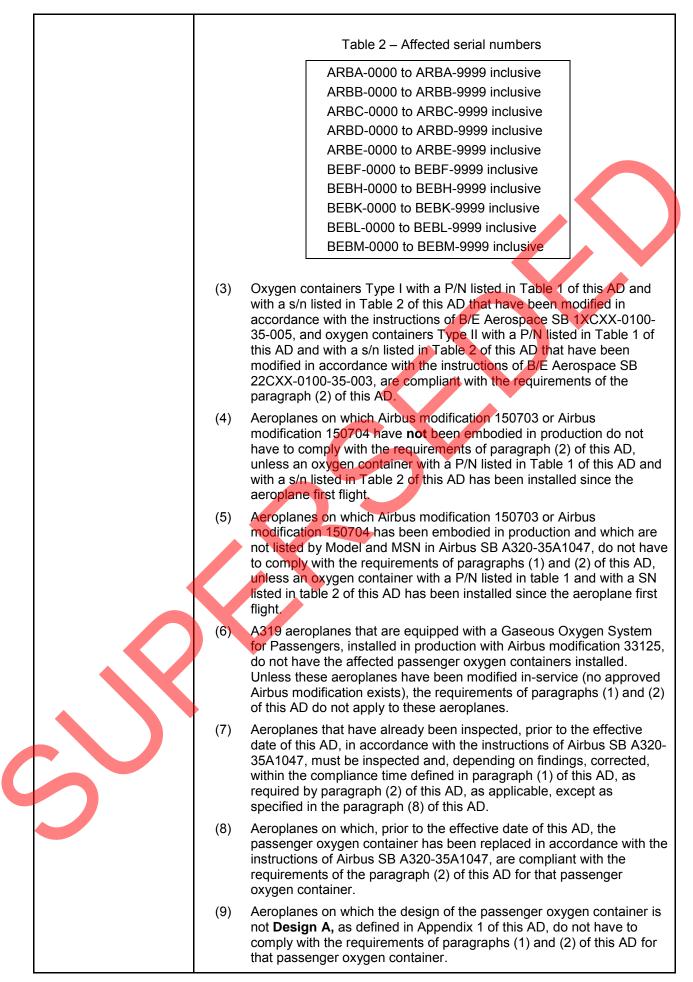
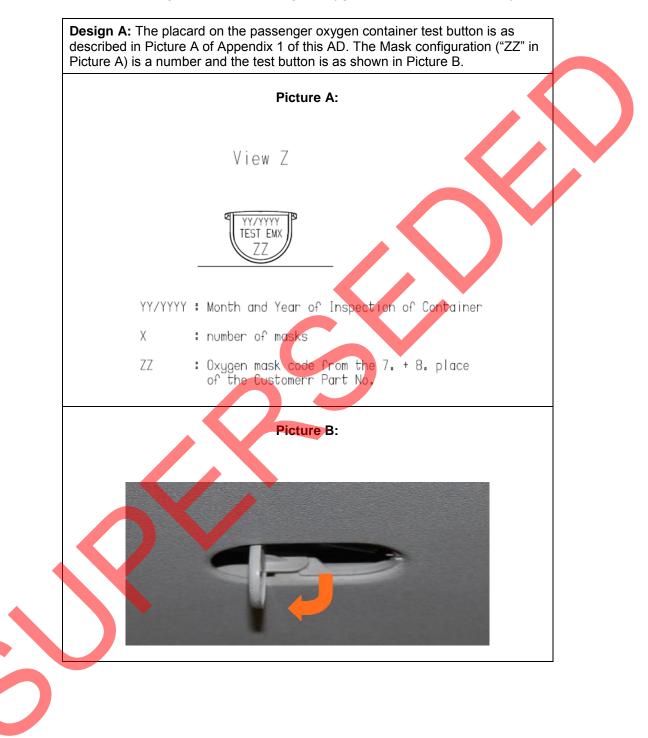
EASA	AIRWORTHINESS DIRECTIVE	
	AD No.: 2012-0083	
	Date: 16 May 2012	
C.	Regulation (EC) No 216/20	Directive (AD) is issued by EASA, acting in accordance with 08 on behalf of the European Community, its Member States and tries that participate in the activities of EASA under Article 66 of
continuing airworthiness of an an aircraft to which an AD app	aircraft shall be ensured by accompli lies, except in accordance with the re	3. In accordance with EC 2042/2003 Annex I, Part M.A.301, the ishing any applicable ADs. Consequently, no person may operate equirements of that AD unless otherwise specified by the Agency of the State of Registry [EC 216/2008, Article 14(4) exemption].
Type Approval Holder's Name :		Type/Model designation(s) :
AIRBUS		A318, A319, A320 and A321 aeroplanes
TCDS Number:	EASA.A.064	
Foreign AD:	Not applicable	
Supersedure:	This AD supersedes EASA	AD 2011-0167 dated 06 September 2011.
ATA 35	Oxygen – Chemical I Identification / Modif	Emergency Oxygen Containers – ication
Manufacturer(s):	Airbus (formerly Airbus Industrie)	
Applicability:	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-111, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all Manufacturer Serial Numbers (MSN).	
Reason:	During production of passenger oxygen containers, the manufacturer B/E Aerospace detected some silicon particles inside the oxygen generator manifolds. Investigation revealed that those particles (chips) had chafed from the mask hoses during installation onto the generator outlets. It was discovered that a defective mask hose installation device had caused the chafing.	
U		ected and corrected, could reduce or block the resulting in injury to passengers when oxygen
	require the identification assemblies. That AD als	I unsafe condition, EASA issued AD 2011-0167 to and modification of the affected oxygen container o prohibited the installation of the affected lane as replacement parts.
	A318-122 were missing	d, it was established that the Models A318-121 and from the Applicability of the AD, and clarification was affected containers, which are only those marked

	B/E Aerospace Systems on the equipment da	ta plate.	
	For the reasons described above, this AD reta AD 2011-0167, which is superseded, expands two aeroplane models, and provides clarity by passenger oxygen containers.	the Applicability by ad	
Effective Date:	30 May 2012		
Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously:		
	(1) Before the accumulation of 5 000 flight cycles, or 7 500 flight hour 24 months, whichever occurs first after 20 September 2011 [the effective date of EASA AD 2011-0167], identify the Part Number and serial number (s/n) of each passenger oxygen container insta on the aeroplane. A review of aeroplane maintenance records is acceptable to accomplish the identification as required by this paragraph, provided those records can be relied upon for that purpose.		
	<ul> <li>(2) If the P/N <u>and</u> s/n of a container, identified as required by parag (1) of this AD, are listed, respectively, in Table 1 (P/N) and Table (s/n) of this AD, accomplish the following actions concurrently, we the compliance time as specified in paragraph (1) of this AD, in accordance with the instructions of Airbus Service Bulletin (SB) 35A1047:</li> </ul>		
	(2.1) Replace the oxygen generator man container with a serviceable man		
	(2.2) Perform an operational check of t	he manual mask releas	
	(2.3) Check if the P/N of the container is listed in B/E Aerospa 1XCXX-0100-35-005 Revision 01, or B/E Aerospace SB 22CXX-0100-35-003 Revision 01, and if is not listed, cor Airbus for instructions and accomplish those instructions accordingly.		
		plish those instructions	
	accordingly.	Numbers	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu	Numbers	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu Type I: 15 and 22 minutes Type II:	Numbers merical value)	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu Type I: 15 and 22 minutes Type II: 12C15Lxxxxx0100 22C	Numbers merical value) 15 and 22 minutes	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu Type I: 15 and 22 minutes Type II: 12C15Lxxxxx0100 22C 12C15Rxxxxx0100 22C	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu Type I: 15 and 22 minutes Type II: 12C15Lxxxxx0100 22C 12C15Rxxxxx0100 22C 13C15Lxxxxx0100 22C	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu Type I: 15 and 22 minutes Type II: 12C15Lxxxxx0100 22C 12C15Rxxxxx0100 22C 13C15Lxxxxx0100 22C	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	
	accordingly. Table 1 – Affected Part (xxxx stands for any alphanu Type I: 15 and 22 minutes Type II: 12C15Lxxxxx0100 22C 12C15Rxxxx0100 22C 13C15Lxxxxx0100 22C 22C	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	
	accordingly.Table 1 – Affected Part(xxxx stands for any alphanuType I: 15 and 22 minutesType II:12C15Lxxxxx010022C12C15Rxxxx010022C13C15Lxxxx010022C13C15Lxxxx010022C13C15Lxxxx010022C14C15Lxxxx010022C	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	
	accordingly.           Table 1 – Affected Part (xxxx stands for any alphanu           Type I: 15 and 22 minutes         Type II:           12C15Lxxxxx0100         22C           12C15Rxxxx0100         22C           13C15Lxxxx0100         22C           13C15Lxxxx0100         22C           13C15Lxxxx0100         22C           13C15Lxxxxx0100         22C           13C15Rxxxx0100         22C           13C15Rxxxx0100         22C           14C15Rxxxx0100         22C	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	
	accordingly.           Table 1 – Affected Part (xxxx stands for any alphanut (xxxx stands for any alphanut)           Type I: 15 and 22 minutes         Type II:           12C15Lxxxxx0100         22C           12C15Exxxxx0100         22C           13C15Exxxxx0100         22C           13C15Exxxxx0100         22C           13C15Exxxxx0100         22C           13C15Exxxxx0100         22C           13C15Exxxxx0100         22C           14C15Exxxxx0100         22C           14C15Exxxxx0100         12C22Lxxxxx0100	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	
	accordingly.           Table 1 – Affected Part (xxxx stands for any alphanu           Type I: 15 and 22 minutes         Type II:           12C15Lxxxxx0100         22C           12C15Lxxxxx0100         22C           13C15Lxxxx0100         22C           13C15Lxxxx0100         22C           13C15Lxxxx0100         22C           13C15Lxxxx0100         22C           13C15Rxxxx0100         22C           13C15Rxxxx0100         22C           13C15Rxxxx0100         22C           13C15Rxxxx0100         22C           14C15Lxxxx0100         22C           14C15Rxxxx0100         12C22Lxxxx0100           12C22Rxxxx0100         12C22Rxxxx0100	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	
	accordingly.           Table 1 – Affected Part (xxxx stands for any alphanu           Type I: 15 and 22 minutes         Type II:           12C15Lxxxxx0100         22C           12C15Exxxxx0100         22C           13C15Exxxxx0100         22C           14C15Exxxxx0100         22C           13C22Exxxxx0100         13C22Exxxxx0100	Numbers merical value) <b>15 and 22 minutes</b> 15Lxxxxx0100 15Rxxxxx0100 22Lxxxxx0100	



	(10) From 20 September 2011 [the effective date of EASA AD 2011-0167], do not install on any aeroplane an oxygen container with a P/N listed in Table1 of this AD <u>and</u> with a s/n listed in Table 2 of this AD, unless the container has been modified in accordance with the instructions of Airbus SB A320-35A1047, or B/E Aerospace SB 1XCXX-0100-35-005, or B/E Aerospace SB 22CXX-0100-35-003, as applicable.
Ref. Publications:	<ul> <li>Airbus SB A320-35A1047 original issue dated 29 March 2011.</li> <li>B/E Aerospace SB 1XCXX-0100-35-005 original issue dated 14 March 2011 or Revision 01 dated 15 December 2011.</li> <li>B/E Aerospace SB 22CXX-0100-35-003 original issue dated 17 March 2011 or Revision 01 dated 20 December 2011.</li> <li>The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.</li> </ul>
Remarks :	<ol> <li>If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> <li>This AD was posted on 23 February 2012 as PAD 12-018 for consultation until 22 March 2012. The Comment Response Document can be found at <u>http://ad.easa.europa.eu/</u>.</li> <li>Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: <u>ADs@easa.europa.eu</u>.</li> <li>For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS, Fax +33 5 61 93 44 51, E-mail: <u>account.airworth-eas@airbus.com</u>.</li> </ol>



Appendix 1 – Design A of the Passenger Oxygen Containers affected by this AD