EASA	AIRWORTHINESS DIRECTIVE
	AD No.: 2013-0147 **Date: 16 July 2013** Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Design Approval Holder's Name:	**Type/Model designation(s):**
AIRBUS	A318, A319, A320 and A321 aeroplanes

TCDS Number:	EASA.A.064
Foreign AD:	Not applicable
Supersedure:	This AD supersedes EASA AD 2010-0071R1 dated 28 May 2010, and EASA AD 2011-0231 dated 09 December 2011.
ATA 05	**Time Limits and Maintenance Checks – Damage Tolerant Airworthiness Limitation Items – ALS Part 2– Amendment**
Manufacturer(s):	Airbus (formerly Airbus Industrie)
Applicability:	Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-111, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers.
Reason:	The airworthiness limitations for Airbus A320 family aeroplanes are currently included in Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) document. The airworthiness limitations applicable to the Damage Tolerant Airworthiness Limitation Items (DT ALI) are currently given in Airbus ALS Part 2, which is approved by EASA. Previously, EASA issued AD 2010-0071R1, which required the implementation of the DT ALI maintenance instructions as specified in Airbus A318/A319/A320/A321 ALI Document ref. AI/SE-M4/95A.0252/96 issue 10 and Airbus A319 Corporate Jet ALI Document ref. AI/SE-M2/95A.1038/99. The new Airbus A318/A319/A320/A321 ALS Part 2 Revision 02, which includes also Airbus A319 Corporate Jet, introduces more restrictive DT ALI maintenance instructions. Failure to comply with these instructions could result in an unsafe condition. Application of new DT ALI tasks 531129-02-2 and 531129-02-3 introduces initial and repetitive inspections of the windshield central lower node continuity fittings, replacing the one time inspection for that subject, previously required by EASA AD 2011-0231.

	For the reasons described above, this AD retains the requirements of EASA AD 2010-0071R1 and those of EASA AD 2011-0231, which are superseded, and requires compliance with all maintenance tasks as described in Airbus A318/A319/A320/A321 ALS Part 2 at Revision 02.
Effective Date:	30 July 2013
Required Action(s) and Compliance Time(s):	Required as indicated, unless accomplished previously: (1) After the effective date of this AD, within the compliance times defined in the Record of Revisions (ROR) pages of the Airbus A318/A319/A320/A321 ALS Part 2 Revision 02, accomplish all applicable maintenance tasks, depending on aeroplane configuration, as specified in Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 02. (2) If, during accomplishment of any task as required by paragraph (1) of this AD, discrepancies are detected, within the applicable compliance time specified in the Airbus maintenance documentation, accomplish the applicable corrective action(s) in accordance with instructions specified (or incorporated by reference) in the applicable Airbus maintenance documentation. If no compliance time is identified in the Airbus maintenance documentation, before next flight, accomplish the corrective action(s) or contact Airbus for approved instructions and within the compliance time specified therein, accomplish those instructions accordingly. If any of the discrepancies, detected during accomplishment of any task as required by paragraph (1) of this AD, is not defined in the Airbus maintenance documentation, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly. (3) Compliance with the requirements of paragraph (1) of this AD can be demonstrated by: (3.1) Revising as follows, the approved Aircraft Maintenance Programme (AMP) on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane: Incorporate all DT ALI maintenance instructions specified in Airbus A318/A319/A320/A321 ALS Part 2 Revision 02. and (3.2) Complying with the approved AMP described in paragraph (3.1) of this AD.
Ref. Publications:	Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2 Revision 02, approved by EASA on 28 May 2013. The use of later approved variations or revisions of this document is acceptable for compliance with the requirements of this AD.
Remarks:	1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 2. This AD was posted on 05 February 2013 as PAD 13-028 for consultation until 05 March 2013. The Comment Response Document can be found at http://ad.easa.europa.eu. 3. Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu. 4. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS - Airworthiness Office - EIAS, Fax +33 5 61 93 44 51, E-mail: account.airworth-eas@airbus.com.