EASA AD No.: 2014-0067

AD No.: 2014-0067 Date: 18 March 2014 Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry IEC 216/2008. Article 14(4) exemption].

2042/2003 Annex I, Part M.A	a.303] or agreed with the Authority of the	State of Registry [EC 216/2008, Article 14(4) exemption].			
Design Approva	l Holder's Name:	Type/Model designation(s):			
FIBERGLAS-TECHNIK RUDOLF LINDNER GmbH & Co. KG		GROB G 102 and G 103 sailplanes and powered sailplanes			
TCDS Number:	EASA.A.250				
Foreign AD:	Not applicable				
Supersedure:	None				
ATA 27	Flight Controls – Rudder Control Unit Cable Pulleys – Inspection / Replacement				
Manufacturer(s):	GROB Werke GmbH & Co KG (formerly Burkhart Grob Flugzeugbau, Burkhardt Grob Luft- und Raumfahrt GmbH & Co. KG)				
Applicability:	GROB G 102 Standard Astir III (with letter "S"), G 102 Club Astir III (with letter "C"), and G 102 Club Astir IIIb, (with letter "Cb") sailplanes, serial numbers (s/n) 5501 to 5652 inclusive.				
	GROB G 103 TWIN II and G 103 A TWIN II ACRO (with letter "K") sailplanes, s/n 3730 to 34078 inclusive.				
	GROB G 103 C TWIN III ACRO sailplanes, s/n 34101 to 34203 inclusive.				
	GROB G 103 C TWIN III sailplanes, s/n 36001 to 36014 inclusive.				
	GROB G 103 C TWIN III SL powered sailplanes, s/n 35002 to 35051 inclusive.				
Reason:	Control cable pulleys made from plastic (white or brown material) in the rudder control unit were reported to develop cracks due to aging. In one case, jamming of the rudder control unit was reported.				
	This condition, if not detected and corrected, could cause cable pulleys to break, potentially jamming the rudder control unit and resulting in loss of control of the sailplane.				
	To address this potential unsafe condition, Fiberglas-Technik issued Technische Mitteilung/Service Bulletin TM-G05/SB-G05 and Anweisung/ Instructions A/I-G05 (one document) to provide instructions for the replacement of plastic cable pulleys with pulleys made from aluminium.				

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	For the reason described above, this AD requires identification and replacement of plastic cable pulleys in the rudder control unit. Plastic cable pulleys may also be installed in the cable circuits of pedal adjustment and/or tow hook actuation, their replacement is not required by this AD.				
Effective Date:	01 April 2014				
Required Action(s) and Compliance Time(s):	(1) Within the co rudder contro plastic cable	ompliance time defined ol unit and, if plastic ca pulleys with aluminium of Fiberglas-Technik T	ess accomplished previously: time defined in Table 1 of this AD, inspect the d, if plastic cable pulleys are installed, replace the rith aluminium cable pulleys in accordance with the as-Technik TM-G05/SB-G05 and A/I-G05. Compliance time for replacement		
		der control unit guration	Compliance time (after the effective date of this AD)		
	Oper	n cable cage	Within 1 month		
	Close	ed cable cage	Within 3 months		
	Note: All G 103 C (TWIN III, TWIN III ACRO) sailplanes and G 103 C (TWIN III SL) powered sailplanes have closed cable cages. (2) From the effective date of this AD, do not install any plastic control cable pulley in the rudder control unit of a sailplane or powered sailplane.				
Ref. Publication:	Fiberglas-Technik TM-G05/SB-G05 and A/I-G05 original issue, dated 17 January 2014. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.				
Remarks:	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. This AD was posted on 30 January 2014 as PAD 14-027 for consultation until 27 February 2014. The Comment Response Document can be found at http://ad.easa.europa.eu. Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu. For any question concerning the technical content of the requirements in this AD, please contact: E-mail: info@LTB-Lindner.com. 				