


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE	
	PAD No.: 11-121 Date: 15 November 2011 Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.	
	In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.	
Type Approval Holder's Name : AIRBUS		Type/Model designation(s) : A310 aeroplanes
TCDS Number : France TCDS No 145		
Foreign AD : Not applicable		
Supersedure : None		
ATA 71	Engine Mounts – Forward Engine Mount Bolts – Replacement	
Manufacturer(s):	Airbus (formerly Airbus Industrie)	
Applicability:	Airbus model A310-203 and A310-203C aeroplanes, all serial numbers.	
Reason:	<p>Post type-certification analyses performed in the frame of the Extended Service Goal (ESG) exercise, have demonstrated a reduced fatigue life for the side link bolts, centre sway link bolts and thrust link bolts of the General Electric (GE) CF6-80A3 forward engine mounts.</p> <p>This condition, if left uncorrected, could result in a deterioration of the structural integrity of the front engine mount bolts.</p> <p>For the reasons described above, this AD requires replacement of all side link bolts, centre sway link bolts and all thrust link bolts of GE CF6-80A3 powered aeroplanes.</p>	
Effective Date:	[TBD: 14 days after final AD issue date]	
Required Action(s) and Compliance Time(s):	Required as indicated, unless previously accomplished: (1) Within 18 months after the effective date of this AD replace the bolts all side link bolts on left hand (LH) and right hand (RH) side of the engine, centre sway link bolts and thrust link bolts of both engines, Part Number (P/N) 9021M88P01 (and 9021M88P02), 9205M81P01, 9021M88P03 (and 9021M88P04) and PN 9205M82P01 with new bolts (same P/N), in accordance with the Accomplishment Instructions of AIRBUS Service Bulletin (SB) A310-71-2037 original issue.	

	(2) Thereafter, at intervals not exceeding 134 months, replace again all forward engine mount bolts of both engines in accordance with the Accomplishment Instructions of AIRBUS Service Bulletin (SB) A310-71-2037 original issue.
Ref. Publications:	Airbus SB A310-71-2037 original issue dated 30 September 2011. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.
Remarks :	<ol style="list-style-type: none"> 1. This Proposed AD will be closed for consultation on 13 December 2011. 2. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail ADs@easa.europa.eu. 3. For any questions concerning the technical content of the requirements in this PAD, please contact: AIRBUS SAS – EAW (Airworthiness Office, Telephone: + 33 5 61 18 41 39, Fax: + 33 5 61 93 44 51).