


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE	
	<p>PAD No.: 13-137</p> <p>Date: 11 September 2013</p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>	
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>		
<p>Design Approval Holder's Name: AIRBUS</p>	<p>Type/Model designation(s): A300, A300-600, A300-600ST and A310 aeroplanes</p>	
TCDS Number:	France n° 145 and EASA.A.014	
Foreign AD:	Not applicable	
Supersedure:	This AD supersedes EASA AD 2007-0293 dated 29 November 2007.	
ATA 05	Time limits / Maintenance checks – Safe life Airworthiness Limitation Items (ALS Part 1) – Amendment	
Manufacturer(s):	Airbus (formerly Airbus Industrie)	
Applicability:	<p>Airbus A300 aeroplanes, all certified models, all Manufacturer Serial Numbers (MSN),</p> <p>Airbus A300-600 aeroplanes, all certified models, all MSN,</p> <p>Airbus A300F4-608ST aeroplanes, all MSN,</p> <p>Airbus A310 aeroplanes, all certified models all MSN.</p>	
Reason:	<p>The airworthiness limitations are currently distributed in the Airbus A300 Airworthiness Limitation Section (ALS), A300-600 ALS and A310 ALS.</p> <p>The instructions and airworthiness limitations applicable to the Safe Life Airworthiness Limitation Items (SL ALI) are given in Airbus A300 ALS Part 1, A300-600 ALS Part 1 and A310 ALS Part 1, which are approved by the European Aviation Safety Agency (EASA).</p> <p>The revision 01 of Airbus A300 ALS Part 1, A300-600 ALS Part 1 and A310 ALS Part 1 introduces more restrictive instructions and/or airworthiness limitations. Failure to comply with the maintenance requirements contained in these revisions could constitute an unsafe condition.</p> <p>For the reasons described above, this AD retains the requirements of EASA AD 2007-0293, which is superseded, and requires the implementation of the new or more restrictive maintenance requirements and/or airworthiness limitations as specified in the Airbus A300 ALS Part 1 Revision 1, A300-600 ALS Part 1 Revision 1 and A310 ALS Part 1 Revision 1, except for the A300 life limits</p>	

	applicable to Main Landing Gear (MLG) barrel assembly, retraction actuator assembly linkage assembly and to flanged duct, for which applicable life limits are defined in EASA AD 2013-0210.
Effective Date:	[TBD: 14 days after final AD issue date]
Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <p>(1) From the effective date of this AD, accomplish the following actions, as specified in the Airbus A300 ALS Part 1 revision 01, Airbus A300-600 ALS Part 1 revision 01, or Airbus A310 ALS Part 1 revision 01, as applicable to aeroplane model and depending on aeroplane configuration:</p> <p>(1.1) Replace each component before or upon reaching the applicable life limit.</p> <p>(1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.</p> <p>(2) Compliance with the requirements of paragraph (1) of this AD can be demonstrated by:</p> <p>(2.1) Revising as follows, the approved Aircraft Maintenance Program (AMP) on the basis of which the Operator or the Owner ensures the continuing airworthiness of each operated aeroplane:</p> <p>Incorporate all maintenance tasks and applicable airworthiness limitations specified in the Airbus A300 ALS Part 1 revision 01, Airbus A300-600 ALS Part 1 revision 01 and Airbus A310 ALS Part 1 revision 01, as applicable to aeroplane model and relevant to the aeroplane configuration and weight variant,</p> <p>and</p> <p>(2.2) Complying with the approved AMP described in paragraph (2.1) of this AD.</p>
Ref. Publications:	<p>Airbus A300 ALS Part 1 Revision 01 approved by EASA on 05 September 2013.</p> <p>Airbus A300-600 ALS Part 1 Revision 01 approved by EASA on 05 September 2013.</p> <p>Airbus A310 ALS Part 1 Revision 01 approved by EASA on 05 September 2013.</p> <p>The use of later approved variations or revisions of these documents is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. This Proposed AD will be closed for consultation on 02 October 2013. 2. Enquiries regarding this PAD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu. 3. For any question concerning the technical content of the requirements in this PAD, please contact: AIRBUS SAS – EIAW (Airworthiness Office), Telephone: + 33 5 61 18 41 39, Fax: + 33 5 61 93 44 51.