

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref. : 84-154-65(B)
In case of any difficulty, reference should be made to the French original issue.

AIRBUS INDUSTRIE

A300 airplanes

Fuselage - Frame 47

The present Airworthiness Directive applies to AIRBUS INDUSTRIE A300 aircraft models A 300 B1, A300 B2-1A, B2-1C, B2K-3C, B2-202 and B2-203 (all serial numbers).

Due to the discovery of a crack at fuselage frame 47 on an aircraft in service and in order to prevent fatigue crack development, compliance with the following is hereby rendered mandatory, unless already accomplished :

- 1/ - Visually inspect fuselage frame 47 in accordance with instructions given in AIRBUS INDUSTRIE Alert Service Bulletin n° 53A193.
This inspection must be performed prior to accumulation of 5000 flights or within the next 300 flights after March 21, 1984, whichever occurs later.
- 2/ - In case no cracks are evidenced during inspection prescribed under paragraph 1 above, no further action is required under the present Airworthiness Directive.
- 3/ - In case a crack is evidenced during inspection prescribed under paragraph 1 above and if crack length is 7 mm or less, repeat the same inspection or perform repair of frame 47 in accordance with instructions given in AIRBUS INDUSTRIE Alert Service Bulletin n° 53A193.
- 4/ - In case a crack is evidenced during inspections prescribed under paragraphs 1 and 3 above and if crack length is greater than 7 mm, perform repair of frame 47 in accordance with instructions given in AIRBUS INDUSTRIE Alert Service Bulletin n° 53A193 within the following times :
 - Within the next 50 flights after crack discovery if crack length is comprised between 7 and 50 mm.
 - Prior to any further flight after crack discovery if crack length is 50 mm or more.

Ref. : AIRBUS INDUSTRIE Alert Service Bulletin N° 53A193
The requirements of this Airworthiness Directive are covered in FRG by LTA N° 84-40/2.

Cancels and Supersedes A.D. n° 84-40-60(B) dated March 14, 1984.

EFFECTIVE DATE : NOVEMBER 14, 1984

v/K

Nov. 7, 1984

AIRBUS INDUSTRIE
A300 airplanes

84-154-65(B)