

AIRWORTHINESS DIRECTIVE

released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref. : 1999-087(A)
In case of any difficulty, reference should be made to the French original issue.

REIMS AVIATION

F406 Airplanes

Fuselage – Wing assembly

1. EFFECTIVITY

REIMS AVIATION F406 airplanes S/N F406-0001 thru F406-0083.

2. ACTIONS

To inspect the canted ribs upper cap in the center wing carry-through area for cracks, in accordance to Service Bulletin REIMS AVIATION No. F406-38, before the next flight.

3. COMPLIANCE

3.1. If a crack is found, measure the length as described in SB F406-38.

- a) If crack is less than 2 inches in length, stop drill the crack using a 0,16 inch drill bit and install doublers in accordance to SB F406-38 within 100 hours T.I.S.
- b) If crack is more than 2 inches in length, install doublers before the next flight .

3.2. If cracks are not found, repeat the inspection each 200 hours T.I.S. or 12 months until accumulating 600 hours T.I.S., or 12 months after the first inspection, or until doublers are installed in accordance to SB F406-38 (whichever occurs first).

Record the application of the present Airworthiness Directive on the aircraft log book.

REF. : Service Bulletin REIMS AVIATION F406-38
CESSNA CAB 98-16

EFFECTIVE DATE : MARCH 06, 1999