	AIRWORTHINESS DIRECTIVE		Distribution:	Issue date:	Page :
	No F-2000-038-032 R1		B	January 21, 2004	1/2
Direction générale de l'aviation civile France	This Airworthiness Directive is published by the DGAC:			<i>Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.</i>	
	<input type="checkbox"/> on behalf of EASA, the Primary Airworthiness Authority for the affected product. <input checked="" type="checkbox"/> as the Registration Airworthiness Authority for the affected aircraft..				
GSAC publication	No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.				
Corresponding foreign Airworthiness Directive(s): Not applicable			Airworthiness Directive(s) replaced: 2000-038-032 original issue		
Person in charge of airworthiness: AIRBUS			Type(s): A300-600ST BELUGA aircraft		
Type certificate(s) No. None TCDS No None					
ATA chapter: 53	Subject: Fuselage - Aft fittings of frame FR40				

1. APPLICABILITY:

AIRBUS A300-600ST aircraft, all certified serial numbers.

2. REASONS:


This Airworthiness Directive is issued in order to prevent propagation of cracks on the aft fittings of frame FR40 at stringers STGR 33 L/H and R/H of the fuselage, which would affect the structural integrity of the airframe.

Revision 1 of this Airworthiness Directive (AD) follows the issue of AIRBUS Service Bulletin (SB) A300-53-9017 specific to the BELUGA aircraft and equivalent to SB A300-53-6048.

3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

The following measures are rendered mandatory from the effective date of this AD at original issue:

- a) On aircraft on which AIRBUS modification No. 10430 has not been embodied (MSN 0001 aircraft at the issuing date of this AD):
- modify the profile of frame FR40 aft fittings and perform a High Frequency Eddy Current (HFEC) inspection, in accordance with the thresholds and the instructions of SB A300-53-9017,
 - depending on the results of the first inspection, repeat the inspection once, in accordance with the threshold defined by SB A300-53-9017,
 - depending on the results of the preceding inspection, repeat the inspection in accordance with the intervals and the instructions of SB A300-53-9017.

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b) On aircraft on which AIRBUS modification No. 10430 has been embodied (MSN 0002, 0003, 0004 and 0005 at the issuing date of this AD):

- perform a High Frequency Eddy Current (HFEC) inspection of frame FR40, in accordance with the thresholds and the instructions of SB A300-53-9017,
- depending on the results of the preceding inspection, repeat this inspection in accordance with the intervals and the instructions of SB A300-53-9017.

Note: Inform AIRBUS of all findings whatever they may be.

4. REFERENCE PUBLICATION:

AIRBUS Service Bulletin A300-53-9017
Any later approved revision of this SB is acceptable.

5. EFFECTIVE DATES:

Original issue : February 05, 2000
Revision 1 : January 31, 2004.

6. REMARK:

For any additional question concerning the technical content of the requirements of this AD, please contact:

AIRBUS – Didier Auriche – Fax : 33 5 61 93 45 80.

7. APPROVAL:

This AD is DGAC approved.

SUPERSEDED