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#### DEPARTMENT OF TRANSPORTATION

**Federal Aviation Administration** 

**14 CFR Part 39** 

[Docket No. FAA-2004-19540; Directorate Identifier 2004-NM-110-AD; Amendment 39-14258; AD 2005-18-18]

RIN 2120-AA64

**Airworthiness Directives; Boeing Model 757 Airplanes** 

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

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**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Boeing Model 757 airplanes. This AD requires inspections of certain wire bundles in the left and right engine-to-wing aft fairings for discrepancies; installation of back-to-back p-clamps between the wire and hydraulic supply tube at the aft end of the right-hand strut only; and associated re-routing of the wire bundles, if necessary. This AD results from a report indicating that a circuit breaker for the fuel shutoff valve tripped due to a wire that chafed against the structure in the flammable leakage zone of the aft fairing, causing a short circuit. We are issuing this AD to prevent chafing between the wire bundle and the structure of the aft fairing, which could result in electrical arcing and subsequent ignition of flammable vapors and possible uncontrollable fire.

**DATES:** This AD becomes effective October 14, 2005.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of October 14, 2005.

**ADDRESSES:** You may examine the AD docket on the Internet at *http://dms.dot.gov* or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for service information identified in this AD.

**FOR FURTHER INFORMATION CONTACT:** Thomas Thorson, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 917-6508; fax (425) 917-6590.

#### **SUPPLEMENTARY INFORMATION:**

#### **Examining the Docket**

You may examine the AD docket on the Internet at *http://dms.dot.gov* or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the ADDRESSES section.

#### Discussion

The FAA issued a supplemental notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Boeing Model 757 airplanes. That supplemental NPRM was published in the Federal Register on July 6, 2005 (70 FR 38823). That supplemental NPRM proposed to require inspections of certain wire bundles in the left and right engine-to-wing aft fairings for discrepancies; installation of back-to-back p-clamps between the wire and hydraulic supply tube at the aft end of the right-hand strut only; and associated re-routing of the wire bundles, if necessary.

#### **Comments**

We provided the public the opportunity to participate in the development of this AD. We have considered the comments received. The commenters support the supplemental NPRM.

#### **Conclusion**

We have carefully reviewed the available data, including the comments received, and determined that air safety and the public interest require adopting the AD as proposed in the supplemental NPRM.

### **Costs of Compliance**

There are about 618 airplanes of the affected design in the worldwide fleet. This AD will affect about 342 airplanes of U.S. registry. The actions will take between 16 and 44 work hours per airplane, depending on airplane configuration, at an average labor rate of \$65 per work hour. Required parts will cost about \$600 per airplane. Based on these figures, the estimated cost of this AD on U.S. operators is between \$560,880 and \$1,183,320, or between \$1,640 and \$3,460 per airplane.

### **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### **Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the ADDRESSES section for a location to examine the regulatory evaluation.

## List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

# **Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39-AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

# **AIRWORTHINESS DIRECTIVE**



Aircraft Certification Service Washington, DC

U.S. Department of Transportation Federal Aviation Administration

We post ADs on the internet at www.faa.gov/aircraft/safety/alerts/

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Title 14 of the Code of Federal Regulations (14 CFR) part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference 14 CFR part 39, subpart 39.3).

**2005-18-18 Boeing:** Amendment 39-14258. Docket No. FAA-2004-19540; Directorate Identifier 2004-NM-110-AD.

#### **Effective Date**

(a) This AD becomes effective October 14, 2005.

#### Affected ADs

(b) None.

### **Applicability**

(c) This AD applies to Model 757-200, -200PF, -200CB, and -300 series airplanes; certificated in any category; equipped with Rolls-Royce engines; as identified in Boeing Alert Service Bulletins 757-28A0073 and 757-28A0074, both Revision 1, both dated February 24, 2005.

#### **Unsafe Condition**

(d) This AD was prompted by a report indicating that a circuit breaker for the fuel shutoff valve tripped due to a wire that chafed against the structure in the flammable leakage zone of the aft fairing, causing a short circuit. The FAA is issuing this AD to prevent chafing between the wire bundle and the structure of the aft fairing, which could result in electrical arcing and subsequent ignition of flammable yapors and possible uncontrollable fire.

### **Compliance**

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

# **One-Time** Inspections/Related Investigative and Corrective Actions

- (f) Within 60 months after the effective date of this AD, do the actions required by paragraphs (f)(1) and (f)(2) of this AD.
- (1) Accomplish the detailed inspections for discrepancies of the wire bundles in the left and right engine-to-wing aft fairings, and applicable and related investigative and corrective actions if necessary, as applicable, by doing all the actions specified in the Accomplishment Instructions of the applicable service bulletins listed in Table 1 of this AD. Accomplish any related investigative and corrective actions before further flight in accordance with the applicable service bulletin.

TABLE 1.—AIRPLANE MODELS AND SERVICE BULLETINS

Boeing airplanes	Boeing Alert	Revision	Date
	Service Bulletin	level	
Model 757–200, –200CB, 200PF series airplanes	757–28A0073	Original	November 20, 2003.
Model 757–200, –200CB, 200PF series airplanes	757–28A0073	1	February 24, 2005.
Model 757–300 series airplanes	757–28A0074	Original	November 20, 2003.
Model 757–300 series airplanes	757–28A0074	1	February 24, 2005.

(2) Install back-to-back p-clamps between the wire and hydraulic supply tube at the aft end of the right-hand strut only; and re-route the wire bundles, if necessary, by doing all the applicable actions specified in the Accomplishment Instructions of Boeing Alert Service Bulletin 757-28A0073 or 757-28A0074, both Revision 1, both dated February 24, 2005; as applicable.

**Note 1:** For the purposes of this AD, a detailed inspection is: "An intensive examination of a specific item, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at an intensity deemed appropriate. Inspection aids such as mirror, magnifying lenses, etc., may be necessary. Surface cleaning and elaborate procedures may be required."

### **Alternative Methods of Compliance (AMOCs)**

(g) The Manager, Seattle Aircraft Certification Office, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

### **Material Incorporated by Reference**

(h) You must use the applicable service bulletin listed in Table 2 of this AD to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., room PL-401, Nassif Building, Washington, DC; on the Internet at <a href="http://dms.dot.gov">http://dms.dot.gov</a>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to

http://www.archives.gov/federal\_register/code\_of\_federal\_regulations/ibr\_locations.html.

▲ TABLE 2.—MATERIAL INCORPORATED BY REFERENCE

<b>Boeing Alert Service Bulletin</b>	<b>Revision level</b>	Date
757–28A0073	Original	November 20, 2003.
757–28A0073	1	February 24, 2005.
757–28A0074	Original	November 20, 2003.
757–28A0074	1	February 24, 2005.

Issued in Renton, Washington, on August 31, 2005.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 05-17772 Filed 9-8-05; 8:45 am]

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