

EASA	AIRWORTHINESS DIRECTIVE	
	<p>AD No.: 2009-0185</p> <p>Date: 20 August 2009</p> <p>Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>	
<p>This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].</p>		
<p>Type Approval Holder's Name : Agusta S.p.A.</p>		<p>Type/Model designation(s) : AB412/AB412EP helicopters</p>
<p>TCDS Number : ENAC Italy SO/A 375</p>		
<p>Foreign AD : Not applicable</p>		
<p>Supersedure : None</p>		
ATA 53		
Fuselage – Left Upper Cap Angle - Inspection		
<p>Manufacturer(s): Agusta S.p.A.</p>		
<p>Applicability: AB412 and AB412EP helicopters, all serial numbers.</p>		
<p>Reason:</p> <p>Following a report of one failure to the left upper cap angle part number (P/N) 212-030-191-1 occurred on one AB412EP helicopter, the preliminary investigations done by the Type Certificate Holder Agusta have identified an Hi-lock hole to be the origin of the crack.</p> <p>To detect and prevent this unsafe condition, this AD requires repetitive visual inspections of the left upper cap angle and, in case of crack, the accomplishment of the associated corrective actions.</p> <p>The requirements of this AD are considered as interim action. If, as a result of the on-going investigation, a terminating action will be later identified, further mandatory actions might be considered.</p>		
<p>Effective Date: 03 September 2009</p>		
<p>Required action(s) and Compliance Time(s):</p> <p>Required as indicated, unless accomplished previously:</p> <p>(1) Within 25 flight hours (FH) after the effective date of this AD but not later than 30 November 2009, whichever occurs first, and thereafter at intervals not to exceed 100 FH, inspect the left upper cap angle P/N 212-030-191-1 in accordance with the instructions of Agusta Mandatory Bollettino Tecnico 412-128.</p> <p>(2) In case of cracks of the upper cap, before further flight, contact</p>		

	Agusta for further instructions and follow their corrective actions.
Ref. Publications:	<p>Agusta Mandatory Bollettino Tecnico 412-128 dated 17 August 2009.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks:	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. 2. The required actions and the risk allowance have granted the issuance of a Final AD with Request for Comments, postponing the public consultation process after publication. 3. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu . 4. For any question concerning the technical content of the requirements in this AD, please contact: Agusta S.p.A. Customer Support – Via per Tornavento, 15 21019 Somma Lombardo (VA) – Italy; Phone + 39 0331 711222; Fax: + 39 0331 711254; E-mail: absereng@agustawestland.com .

Superseded