EASA AD No.: 2015-0162

## EASA

## **AIRWORTHINESS DIRECTIVE**

AD No.: 2015-0162

**Date: 06 August 2015** 

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EU 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EU 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].			
Design Approval Holder's Name:		Type/Model designation(s):	
TURBOMECA		ARRIEL, 2 engines	
TCDS Number:	EASA.E.001		
Foreign AD:	Not applicable		
Supersedure:	None		
ATA 72	Engine – Accessory Gea Replacement	ar Box Module M01 – Inspection /	
Manufacturer(s):	Turboméca		
Applicability:	ARRIEL 2B, 2B1, 2B1A, 2B1 engines, all serial numbers.	IB, 2C, 2C1, 2C2, 2D, 2E, 2N, 2S1 and 2S2	
	Helicopters AS350B3, EC13	be installed on, but not limited to, Airbus 0B4, EC130T2, AS365N3, EC155B and EC155B1, and MBB-BK 117 D-2 (EC145T2 or H145), Avic orsky S-76C helicopters.	
Reason:	An uncommanded in-flight shut-down (IFSD) of an ARRIEL 2 engine was reported, caused by rupture of the 41-tooth gear, which forms part of the bevel gear of the accessory gearbox (module M01). The subsequent investigation revealed that wear on the housing of the front bearing of this gear was a major contributor to this rupture. In addition, the investigation showed that this wear mechanism had resulted in positive Spectrometric Oil Analysis (SOA) indications before the event.		
		d and corrected, could potentially lead to further llting in an emergency landing.	
		nsafe condition, Turboméca issued Mandatory 72 2861 to provide SOA check instructions.	
		bove, this AD requires repetitive SOA checks and, placement of the module M01.	

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Effective Date:	20 August 2015	
Required Action(s) and Compliance Time(s):	<ul> <li>Required as indicated, unless accomplished previously:</li> <li>(1) Within the compliance time specified in Table 1 of this AD, perform a visual inspection in accordance with the instructions of Turboméca MSB 292 72 2861 to identify, if a machined front casing Part Number (P/N) 0292120650 is installed on that engine.</li> <li>(2) For engines equipped with a module M01 fitted with a machined front casing P/N 0292120650 (as identified in Turboméca MSB 292 72 2861), within the compliance time specified in Table 1 of this AD, and, thereafter, at intervals not to exceed 100 EH, unless otherwise required by paragraph (3) and (4) of this AD, accomplish a SOA check in accordance with the</li> </ul>	
	instructions of Turboméca MSB 292 72 2861.  Table 1 – Initial SOA check	
	Engine Hours (EH) accumulated by the Module M01 (since first installation on an engine or since last overhaul)  Compliance Time	
	Less than 800 EH  Before exceeding 850 EH since first installation of the Module M01 on an engine, or since last Module M01 overhaul, as applicable	
	800 EH or more, or EH not known Within 50 EH after the effective date of this AD	
	If, during any SOA check as required by paragraph (2) of this AD, an aluminium concentration has value between 0.8 ppm and 1.4 ppm (inclusive), accomplish further SOA checks at intervals not to exceed 50 EH to comply with paragraph (2) of this AD.	
	(4) If, during any SOA check, as required by paragraph (2) or (3) of this AD the aluminium concentration is above 1.4 ppm, within 50 EH after that S check, replace the module M01 with a serviceable module M01 (refer to paragraph (5) of this AD), in accordance with the instructions of Turboméca MSB 292 72 2861.	
	(5) For the purpose of this AD, a serviceable module M01 is either a module M01 fitted with a machined front casing P/N 0292120650, that is subjected to SOA checks in accordance with paragraph (2) and (3) of this AD, or a module M01 fitted with a casted front casing P/N 0292127020	
	(6) From the effective date of this AD, it is allowed to install a module M01 fitted with a machined front casing P/N 0292120650, provided it is subjected to SOA checks in accordance with paragraph (2) and (3) of this AD.	
Ref. Publications:	Turboméca MSB 292 72 2861 version A dated 24 April 2015.  The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks:	<ol> <li>If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.</li> </ol>	
	<ol> <li>This AD was posted on 23 June 2015 as PAD 15-079 for consultation until 21 July 2015. No comments were received during the consultation period.</li> </ol>	
	3. Enquiries regarding this AD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail: <a href="mailto:ADs@easa.europa.eu">ADs@easa.europa.eu</a> .	

4. For any question concerning the technical content of the requirements in this AD, please contact:

**Turboméca S.A.**, ARRIEL 2 Customer Support 40220 Tarnos, France

Fax: +33 5 59 74 45 15, or your usual or nearest TURBOMECA technical representative (refer to <a href="http://www.turbomeca-support.com">http://www.turbomeca-support.com</a>).

