



## Airworthiness Directive

**AD No.:** 2019-0267R1

**[Correction: 13 November 2019]**

**Issued:** 12 November 2019

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:** AIRBUS HELICOPTERS  
**Type/Model designation(s):** SA 365, AS 365, EC155 helicopters

**Effective Date:** Revision 1: 19 November 2019  
 Original issue: 28 October 2019

**TCDS Number(s):** EASA.R.105

**Foreign AD:** Not applicable

**Revision:** This AD revises EASA Emergency AD 2019-0267-E dated 25 October 2019.

### ATA 65 – Tail Rotor – Double Bearing – Inspection / Replacement

**Manufacturer(s):**

Airbus Helicopters (AH), formerly Eurocopter, Eurocopter France, Aerospatiale, Sud Aviation

**Applicability:**

SA 365 N1, AS 365 N2, AS 365 N3, EC155 B and EC155 B1 helicopters, all serial numbers (s/n).

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The applicable ASB:** AH Emergency Alert Service Bulletins (ASB), single publication, applicable to the different helicopter types and models as identified in Table 1 of this AD.

Table 1 – Applicable ASB

| Type / Model                    | ASB Number (original issue) |
|---------------------------------|-----------------------------|
| SA 365 N1, AS 365 N1, N2 and N3 | 01.00.71                    |
| EC155 B and B1                  | 04A016                      |

**Affected part:** Double bearing Part Number (P/N) 704A33-651-245 or P/N 704A33-651-246, installed on a tail rotor gearbox (TGB) P/N 365A33-6005-09.



**Serviceable part:** Double bearing P/N 704A33-651-245 or P/N 704A33-651-246 which is new (never previously installed).

**Groups:** Group 1 are helicopters having an affected part which has accumulated less than 500 flight hours (FH) since its first installation on a helicopter. Group 2 are helicopters having an affected part which has accumulated 500 FH or more since its first installation on a helicopter.

**Close monitoring:** Close monitoring procedure in accordance with Work Card 05-53-00-201 (MET) or Aircraft Maintenance Manual (AMM) Task 05-50-01-211.

**Reason:**

Following a regular TGB chip detector check on an AS 365 N2, equipped with a TGB P/N 365A33-6005-09 (terminating action for EASA AD 2017-0125), a certain quantity of abrasion particles and non-critical scale particles have been found, triggering the close monitoring procedure according to the applicable maintenance instructions. Since no anomaly has been reported during the close monitoring regime, the helicopter has returned to the normal inspection program, but after a few flight hours, during an unscheduled check, a large amount of critical scale particles has been found on the chip detector's plug. The particles belong to the double bearing (pitch control rod bearing) installed inside the TGB. The reported event showed a speed of degradation faster than expected.

This condition, if not detected and corrected, could lead to loss of yaw control of the helicopter.

To address this unsafe condition, AH issued the applicable ASB, providing inspection and replacement instructions, and EASA issued Emergency AD 2019-0267-E to require repetitive inspections of the TGB chip detector for particles and replacement of the double bearing.

Since that AD was issued, it was determined that for Group 1 helicopters the threshold for initial replacement of the double bearing can be extended. This AD is revised accordingly.

This AD is still considered an interim action and further AD action may follow.

This AD is republished to correct the number of the EASA AD revised by this AD.

**Required Action(s) and Compliance Time(s):**

Required as indicated, unless accomplished previously:

**Inspection(s):**

- (1) For Group 1 helicopters under close monitoring on 28 October 2019 [the effective date of the original issue of this AD]: Within the applicable compliance time as identified in the close monitoring and until completion of the close monitoring, assess any particles collected during the close monitoring in accordance with the instructions of paragraph 3.B.2 of the applicable ASB.
- (2) For Group 1 helicopters under close monitoring on 28 October 2019 [the effective date of the original issue of this AD]: Upon completion of the close monitoring regime, and, thereafter, at intervals not exceeding 25 flight hours (FH), accomplish TGB chip detector inspection in accordance with the instructions of paragraph 3.B.2 of the applicable ASB.



- (3) For Group 1 helicopters not under close monitoring on 28 October 2019 [the effective date of the original issue of this AD]: Within 25 FH after the effective date of this AD, and thereafter at intervals not exceeding 25 FH, accomplish TGB chip detector inspection in accordance with the instructions of paragraph 3.B.2 of the applicable ASB.
- (4) For Group 2 helicopters under close monitoring on 28 October 2019 [the effective date of the original issue of this AD]: Before next flight after the effective date of this AD, assess any particles collected during the close monitoring in accordance with the instructions of paragraph 3.B.2 of the applicable ASB, or accomplish a double bearing washing in accordance with the instructions of paragraph 3.B.3 of the applicable ASB.
- (5) For Group 2 helicopters (all), unless already accomplished as required by paragraph (4) of this AD, within 15 FH after 28 October 2019 [the effective date of the original issue of this AD], accomplish a double bearing washing in accordance with the instructions of paragraph 3.B.3 of the applicable ASB.
- (6) For Group 2 helicopters: After accomplishment of the double bearing washing as specified in paragraph (4) or as required by paragraph (5) of this AD, accomplish the inspections as required by paragraph (6.1) or (6.2) of this AD (see Note 1 of this AD).
- (6.1) Inspect the TGB chip detector in accordance with the instructions of paragraph 3.B.2 of the applicable ASB at intervals not exceeding 10 FH and accomplish a double bearing washing in accordance with the instructions of paragraph 3.B.3 of the applicable ASB at intervals not exceeding 30 FH.
- (6.2) Inspect the TGB chip detector in accordance with the instructions of paragraph 3.B.2 of the applicable ASB after every last flight of the day or at intervals not exceeding 5 FH, whichever occurs first.

Note 1: After accomplishment of a double bearing washing in accordance with the instructions of paragraph 3.B.3 of the applicable ASB, subsequent inspections can be accomplished as required by paragraph (6.1) or (6.2) of this AD.

**Corrective Action(s):**

- (7) If, during any action as required by paragraphs (1) to (6) of this AD, as applicable, any discrepancy, as identified in the applicable ASB, is detected, accomplish the applicable corrective action(s) in accordance with the instructions of, and within the compliance time as identified in, the applicable ASB.

**Part(s) replacement:**

- (8) For all helicopters: Before an affected part exceeds 610 FH since first installation on a helicopter, or within 110 FH after 28 October 2019 [the effective date of the original issue of this AD], whichever occurs later, and, thereafter, at intervals not to exceed 500 FH, replace that affected part with a serviceable part. This can be accomplished in accordance with the instructions of the applicable AMM (see Note 2 of this AD).



Note 2: When an affected part installed on a Group 1 helicopter exceeds 500 FH since first installation on a helicopter, that helicopter is Group 2, for which relevant requirements of this AD are applicable.

(9) [Cancelled – merged with Paragraph (8) of this AD].

**Terminating Action(s):**

(10) None.

**Parts Installation:**

(11) From 28 October 2019 [the effective date of the original issue of this AD], it is allowed to install on any helicopter a double bearing, provided it is a serviceable part, as defined in this AD, and that, following installation, it is inspected as required by this AD for Group 1 helicopters.

(12) From 28 October 2019 [the effective date of the original issue of this AD], it is allowed to install on any helicopter a TGB P/N 365A33-6005-09, provided it is equipped with a serviceable part, as defined in this AD, and that, following installation, that serviceable part is inspected as required by this AD for Group 1 helicopters.

**Ref. Publications:**

AH AS365 EASB 01.00.71 original issue dated 24 October 2019.

AH EC155 EASB 04A016 original issue dated 24 October 2019.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication.
3. Enquiries regarding this AD should be referred to the EASA Programming and Continued Airworthiness Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#).
5. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters (Technical Support)  
Web portal: <https://keycopter.airbushelicopters.com> Technical Requests Management, or [TechnicalSupport.Helicopters@airbus.com](mailto:TechnicalSupport.Helicopters@airbus.com).

