



## Airworthiness Directive

**AD No.:** 2021-0093

**Issued:** 30 March 2021

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

AIRBUS

**Type/Model designation(s):**

A300-600 aeroplanes

**Effective Date:** 13 April 2021

**TCDS Number(s):** EASA.A.172

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA 05 – Time Limits / Maintenance Checks – Airworthiness Limitations Section Part 2 / Damage Tolerant Airworthiness Limitation Items – Amendment

**Manufacturer(s):**

Airbus, formerly Airbus Industrie

**Applicability:**

Airbus A300-600 aeroplanes, all certified models, all manufacturer serial numbers.

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The Variation:** Airbus A300-600 Airworthiness Limitations Section (ALS) Part 2 Revision 03 Variation 3.3.

**The AMP:** The approved Aircraft Maintenance Programme (AMP) on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane. For Airbus A300-600 aeroplanes operated under EU regulation, compliance with the approved AMP is required by Commission Regulation (EU) [1321/2014](#), Part M.A.301, paragraph 3.

**New and/or more restrictive tasks:** This includes all tasks that are new and all tasks for which a threshold or interval was reduced, which were introduced through the Variation (as defined in this AD) since the previous ALS Revision that is currently incorporated in the AMP.



**Reason:**

The airworthiness limitations for Airbus A300-600 aeroplanes, which are approved by EASA, are currently defined and published in the Airbus A300-600 ALS document(s). These instructions have been identified as mandatory for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.

EASA previously issued AD 2019-0090 and AD 2020-0111R2, requiring the actions described in Airbus A300-600 ALS Part 2 Revision 03 and Variations 3.1 and 3.2, respectively.

Since those ADs were issued, Airbus published the Variation, as defined in this AD, which contains new and/or more restrictive tasks. This Variation is expected to be incorporated into Airbus A300-600 ALS Part 2 at the next revision.

For the reason described above, this AD requires accomplishment of the actions specified in the Variation. EASA AD 2019-0090 and AD 2020-0111R2 are not superseded by this AD.

**Required Action(s) and Compliance Time(s):**

Required as indicated, unless accomplished previously:

**Maintenance Tasks:**

- (1) From the effective date of this AD, within the thresholds and intervals as defined in the Variation, accomplish all maintenance tasks as specified in the Variation, as applicable to aeroplane model and depending on aeroplane configuration.

Where this AD requires a task which is already required per EASA AD 2019-0090 and/or EASA AD 2020-0111R2, the instructions of the Variation invalidate the instructions of Airbus A300-600 ALS Part 2 Revision 03 and/or Variation 3.1 or 3.2.

**Corrective Action(s):**

- (2) In case of finding discrepancies during accomplishment of any task as required by paragraph (1) of this AD, before next flight, accomplish the applicable corrective action(s) in accordance with the applicable Airbus maintenance documentation. If a detected discrepancy cannot be corrected by using existing Airbus instructions, before next flight, contact Airbus for approved instructions and accomplish those instructions accordingly.

**AMP Revision:**

- (3) Within 12 months after the effective date of this AD, revise the approved AMP by incorporating the tasks and associated thresholds and intervals described in the Variation, as applicable to aeroplane model and depending on aeroplane configuration.

**Recording AD Compliance:**

- (4) When the AMP of an aeroplane has been revised as required by paragraph (3) of this AD, that action ensures continued accomplishment of the tasks as required by paragraph (1) of this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) of this AD, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.



**Ref. Publications:**

Airbus A300-600 ALS Part 2 Revision 03 Variation 3.3 dated 12 January 2021.

The use of later approved revisions of the above-mentioned document, or of an ALS revision which includes the technical content of the Variation, is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 24 February 2021 as PAD 21-031 for consultation until 24 March 2021. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – IIAW (Airworthiness Office)  
E-mail: [continued.airworthiness-wb.external@airbus.com](mailto:continued.airworthiness-wb.external@airbus.com).

