

Airworthiness Directive

AD No.: 2023-0056

Issued: 16 March 2023

Note: This Airworthiness Directive (AD) is issued by EASA, action recordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member 2010, but has been third countries that participate in the activities of EASA under April 2012 of the Begulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1. (2014 Anne) Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing an endiced by ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of the specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or age 2d with a Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS S.A.S.

Effective Date: 30 March 2023

TCDS Numbers: EASA.A.004, EASA.A.015

Foreign AD: Not applicable

Supersedure: None

 Type/n
 del &
 Specton(s):

 A330 at a AS
 aeroplanes

ATA 49 – Airborne Auxiliary Power – A Replacement

Poyler Unit Scroll Housing Assembly –

Manufacturer(s):

Airbus, formerly Airbus Industrie

Applicability:

Airbus A330-201, A330-202, A 30-203, 330-223, A330-223F, A330-243, A330-243F, A330-301, A330-302, A330-303, A330-5, A330-3, A330-323, A330-341, A330-342, A330-343, A330-841, A330-941 and A330-743L peroplaneties an manufacturer serial numbers (MSN); and

Airbus A340-211, A340-212, A34, 13, A340-311, A340-312 and A340-313 aeroplanes, all MSN.

Definitions

For the process of the following definitions apply:

The SB: How ell Service Bulletin (SB) 5053-181-49-7895.

Affected part: Auximary Power Unit (APU) Scroll Housing Assemblies, having Part Number (P/N) 5053-181-001-501 or P/N 0331207990.

SR-1 Affected Part: An affected part marked with 'SR-1' next to its P/N marking.



Serviceable part: APU Scroll Housing Assemblies, having P/N 5053-181-002-501 or P/N 5053-881-002-501.

Affected APU: Honeywell GTCP331-350[C] APUs, having serial number up to P-904 inclusive, that have an affected part installed.

Serviceable APU: Any Honeywell GTCP331-350[C] APU which is not an affected The SB provides instructions to modify an affected APU into a serviceable APU.

Groups: Group 1 aeroplanes are those that have an affected part installed. Group a poplanes are those that do not have an affected part installed.

Reason:

Occurrences have been reported where cracks were found in the scoll here assembly of GTCP331-350 APUs.

This condition, if not corrected, could lead to hot air leakage, which conservent damage to the APU compartment and loss of the APU doors, possibly resulting in damage to the aeroplane.

To address this potential unsafe condition, Honeywell usue, the Se, providing instructions to replace the affected part with a serviceable part, as defined in the AD.

For the reason described above, this AD required replacement of affected APU with serviceable APU, and prohibits (re)installation of affected A

After the publication of the Proposal to issert this AD, it has been determined that additional A330 models, not included in the applicability of the TD, are affected. Additional consultation has not been considered in the interest of affect the prevalue.

Required Action(s) and Compliance

Required as indicated, unless accomplying during iously:

Modification:

- (1) For Group 1 aeroplanes. Eithin 36 roonths after the effective date of this AD, replace the affected APU with a servicea. At J. This can be accomplished in accordance with the instructions of the appropriate Airst seircraft maintenance manual (AMM).
- (2) Re-identification from SR-1 affected part on an aeroplane in accordance with the instructions of the SB is a complete the comply with the requirements of paragraph (1) of this AD for that a coplane.

Part Insta

- (3) Do not install an affected APU on any aeroplane, as required by paragraph (3.1) or (3.2) of this AD, as applicable.
 - (3.1) For Group 1 aeroplanes: After modification of the aeroplane as required by paragraph (1) of this AD, or as specified in paragraph (2) of this AD, as applicable.



(3.2) For Group 2 aeroplanes: From the effective date of this AD.

Ref. Publications:

Honeywell SB 5053-181-49-7895 original issue dated 21 July 2006.

Remarks:

- 1. If requested and appropriately substantiated, EASA can approve Alternative thods of Compliance for this AD.
- 2. This AD was posted on 23 January 2023 as PAD 23-008 for consultation units 25 Forwary 2023. The Comment Response Document can be found in the EASA Safet run Lations Tool, in the compressed (zipped) file attached to the record for this AD.
- 3. Enquiries regarding this AD should be referred to the EASA fety Inc. on Section, Certification Directorate. E-mail: <u>ADs@easa.europa.eu</u>.
- 4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which they occur, or have occurred on a product, part or appliance not affected by this AD can be reported to the <u>EU aviation safety</u> reporting system. This may include reporting on the same or unilar components, other than those covered by the design to which this AD applies of the some unsafe condition can exist or may develop on an aircraft with those components installed under an FAA Parts Manufacturer opproved MA), Supplemental Type Certificate (STC) or other modification.
- 5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS 1IAL (Airworthiness On.) E-mail: <u>airworthiness.A330-A340@airbus.com</u>.

