





Note 1: Unless indicated otherwise, the engine hours and engine cycles specified in Table 1 of this AD are those accumulated by the engine on the effective date of this AD.

**Corrective Action(s):**

- (2) If, during the inspection as required by paragraph (1) of this AD, any discrepancy or damaged affected part is detected, as defined in the NMSB, within the compliance times defined in paragraph '3. Accomplishment Instructions' of the NMSB, as applicable, replace the affected part with a serviceable part in accordance with the instructions of the NMSB.

**Parts Installation:**

- (3) From the effective date of this AD, it is allowed to install an affected part on an engine, provided that the part is a serviceable part.

**Engine Installation:**

- (4) From the effective date of this AD, installation of an engine on an aeroplane is allowed, provided the engine passed an inspection (no finding detected or finding corrected) in accordance with the instructions of the NMSB.

**Reporting:**

- (5) Within 2 days after accomplishment of the inspection, as required by paragraph (1) or (4) of this AD report the inspection result to Rolls-Royce. Using the 'Feedback Sheet NMSB 72-AL165' (Appendix 1 of the NMSB) is an acceptable method to comply with this requirement.

**Ref. Publications:**

Rolls-Royce NMSB 72-AL165 original issue dated 05 September 2024.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. The results of the safety assessment have indicated the need for immediate publication and notification, without the full consultation process.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.



5. For any question concerning the technical content of the requirements in this AD, please contact your designated Rolls-Royce representative, or download the publication from your Rolls-Royce Care account at <https://customers.rolls-royce.com>.

If you do not have a designated representative or Rolls-Royce Care account, please contact **Corporate Communications** at **Rolls-Royce plc**, P.O. Box 31, Derby, DE24 8BJ, United Kingdom Telephone +44 (0)1332 242424 or send an email through <https://www.rolls-royce.com/contact-us/civil-aerospace.aspx> identifying the correspondence as being related to **Airworthiness Directives**.

Superseded

