



## Airworthiness Directive

**AD No.:** 2025-0180

**Issued:** 15 August 2025

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

AIRBUS S.A.S.

**Type/Model designation(s):**

A350 aeroplanes

**Effective Date:** 29 August 2025

**TCDS Number(s):** EASA.A.151

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA 25 – Equipment / Furnishings – Slide Upper Latches – Inspection

**Manufacturer(s):**

Airbus

**Applicability:**

Airbus A350-941 and A350-1041 aeroplanes, all manufacturer serial numbers.

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The AOT:** Airbus Alert Operator Transmission (AOT) A25P033-25.

**MP task:** Maintenance Procedure (MP) Task A350-A-25-62-47-00001-720A-A dated earlier than 04 July 2025.

**Aeroplane date of manufacture:** The date of transfer of title (ownership) of the aeroplane upon delivery by Airbus to the first operator, which is referenced in Airbus documentation.

**Affected part:**

For an aeroplane having a date of manufacture, as defined in this AD, equal to or earlier than 07 November 2024: all slide upper latch assemblies.



For an aeroplane having a date of manufacture later than 07 November 2024: a slide upper latch assembly on which the MP task, as defined in this AD, has been accomplished.

**Groups:** Group 1 aeroplanes are those that have an affected part installed. Group 2 aeroplanes are those that do not have an affected part installed.

**Reason:**

An occurrence was reported where an evacuation slide decorative cover detached in flight, with the slide pack found resting unsecured on the cabin floor. The root cause was attributed to an incorrect installation procedure, potentially leading to installation errors.

This condition, if not detected and corrected, could result in the detachment of the slide pack, possibly preventing timely evacuation in emergency case, leading to injury to aeroplane occupants.

To address this potential unsafe condition, Airbus issued the AOT to provide inspection instructions and updated the “installation of a packed Slide/Raft” instructions.

For the reason described above, this AD requires a one-time visual inspection of the affected parts and, depending on findings, accomplishment of applicable corrective action(s).

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

**Inspection(s):**

- (1) For Group 1 aeroplanes: Within 2 months after the effective date of this AD, accomplish a visual inspection of each affected part in accordance with the instructions of the AOT.

**Corrective Action(s):**

- (2) If, during the inspection as required by paragraph (1) of this AD, discrepancies, as specified in the AOT, are detected, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the AOT.

**Credit:**

- (3) Maintenance actions accomplished on an affected part, in accordance with the instructions of MP Task A350-A-25-62-47-00001-720A-A dated equal or later than 04 July 2025 are acceptable to comply with the requirements of paragraph (1) and (2), as applicable, of this AD for that affected part.

**Additional Maintenance Requirements:**

- (4) For Group 1 and Group 2 aeroplanes: From the effective date of this AD, do not accomplish any maintenance action on an aeroplane in accordance with the instructions of the MP task, as defined in this AD.

**Ref. Publications:**

Airbus AOT A25P033-25 original issue dated 18 June 2025.



The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication. All interested persons may send their comments, referencing the AD Number, to the E-mail address specified in below Remark 3, prior to 12 September 2025. Only if any comment is received during the consultation period, a Comment Response Document will be published in the [EASA Safety Publications Tool](#), in a compressed ('zipped') file, attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS S.A.S. A350 XWB (1IAK), E-mail: [continued-airworthiness.a350@airbus.com](mailto:continued-airworthiness.a350@airbus.com).

