Emergency Airworthiness Directive

AD No.: 2026-0015-E

Issued: 22 January 2026

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS HELICOPTERS

Type/Model designation(s):

AS 332 and EC 225 helicopters

Effective Date:	26 January 2026
TCDS Number(s):	EASA.R.002
Foreign AD:	Not applicable
Supersedure:	None

ATA 55 – Stabilizers – Horizontal Stabilizer – Inspections

Manufacturer(s):

Airbus Helicopters (AH)

Applicability:

AS 332 C, AS 332 C1, AS 332 L, AS 332 L1, AS 332 L2 and EC 225 LP helicopters, all serial numbers.

Definitions:

For the purpose of this AD, the following definitions apply:

The EASB: AH Emergency Alert Service Bulletin (EASB) AS332-05-00-0004 or EASB EC225-05-00-0004, as applicable.

Affected part: Metallic horizontal stabilizer having Part Number (P/N) 332A1310XXXX (manufacturer P/N 332A13-10XX-XX) for AS 332 helicopters, where "XXXX" or "XX-XX" represents any numerical sequence; and metallic horizontal stabilizers, any P/N for EC 225 helicopters.

Groups: Group 1 helicopters are those that have an affected part installed. Group 2 helicopters are those that do not have an affected part installed.

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Reason:
An occurrence was reported of loss of the horizontal stabilizer on an EC 225 LP helicopter during flight.

This condition, if not corrected, could lead to loss of control of the helicopter.

Due to similarity of design, AS 332 helicopters might be also affected.

To address this potential unsafe condition, AH published the EASB, providing instructions for an initial inspection of the affected part, including checking the tightening torque of the nut that holds the tube assembly and the horizontal stabilizer, measurements of the diameter of horizontal stabilizer internal bracket and the end of the tube assembly, and repetitive inspections of the affected part.

For the reason described above, this AD requires initial and repetitive inspections of the affected part, and, depending on findings, accomplishment of applicable corrective action(s).

This AD is considered to be an interim action and further AD action may follow.

Required Action(s) and Compliance Time(s):
Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Inspections:
(1) For Group 1 helicopters: Within 20 flight hours (FH) or 30 days, whichever occurs first after the effective date of this AD, inspect the affected part, including the tube assembly, and perform the measurements in accordance with the instructions of the EASB ('Check with removal of the horizontal stabilizer').

(2) Within 50 FH after the inspection as required by paragraph (1) of this AD, and, thereafter, at intervals not exceeding 50 FH, inspect the affected part in accordance with the instructions of the EASB ('Visual check without removal of the horizontal stabilizer').

Corrective Action(s):
(3) If, during any inspection, as required by paragraph (1) or (2) of this AD, as applicable, any discrepancy, as specified in the EASB, is detected, before next flight, contact AH for approved repair instructions and, within the compliance time specified therein, accomplish those instructions accordingly.

Terminating Action(s):
(4) None.

Part(s) Installation:
(5) For Group 1 and Group 2 helicopters: From the effective date of this AD, it is allowed to install an affected part on a helicopter, provided it is new (not previously installed on any helicopter) or that, concurrently with the installation, it passes an inspection (no discrepancy detected) as

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required by paragraph (1) of this AD. After that installation, the helicopter is effectively a Group 1 and the actions as required by this AD must be accomplished on that helicopter.

Ref. Publications:

AH EASB AS332-05-00-0004 original issue dated 20 January 2026.

AH EASB EC225-05-00-0004 original issue dated 20 January 2026.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.

2. The results of the safety assessment have indicated the need for immediate publication and notification, without the full consultation process.

3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

5. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters (Technical Support) at:
Web portal: https://airbusworld.helicopters.airbus.com / Technical Requests Management, or
E-mail: TechnicalSupport.Helicopters@airbus.com, or Telephone: +33 (0)4 42 859 789.

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