



Airworthiness Directive

AD No.: 2026-0056

Issued: 16 March 2026

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS HELICOPTERS

Type/Model designation(s):

SA 365, AS 365 and EC 155 helicopters

Effective Date: 30 March 2026

TCDS Number(s): EASA.R.105

Foreign AD: Not applicable

Supersedure: None

ATA 71 – Powerplant – Engine Firewalls – Modification

Manufacturer(s):

Airbus Helicopters (AH), formerly Eurocopter (EC), Eurocopter France, Aerospatiale, Sud Aviation

Applicability:

SA 365 N, SA 365 N1, AS 365 N2, AS 365 N3, EC 155 B and EC 155 B1 helicopters, all serial numbers.

Definitions:

For the purpose of this AD, the following definition applies:

The ASB: AH Alert Service Bulletin (ASB) ASB AS365-71-33-0001 or EC155-71-33-0001, as applicable.

Reason:

During the investigation related to an engine compartment fire, some gaps were found between the seals and engines cowling.

This condition, if not corrected, could lead, in case of an engine fire, to fire propagation from one engine compartment to the other and to reduced effectivity of the extinguishing agent, possibly resulting in loss of control of the helicopter.



To address this potential unsafe condition, AH issued the ASB, as defined in this AD, to provide instructions to install angle-brackets along the firewalls of the engine compartment, to improve the sealing of the engine firewalls.

For the reason described above, this AD requires modification of the firewalls of the engine compartment.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Modification:

Within 17 months or 425 flight hours, whichever occurs first after the effective date of this AD, modify the firewalls of the engine compartment in accordance with the instructions of the ASB.

Ref. Publications:

Airbus Helicopters ASB AS365-71-33-0001 Revision 001 dated 01 October 2025.

Airbus Helicopters ASB EC155-71-33-0001 Revision 001 dated 01 October 2025.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 29 January 2026 as PAD 26-017 for consultation until 26 February 2026. The Comment Response Document can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters (Technical Support) – Aéroport de Marseille Provence, 13725 Marignane Cedex, France, Telephone: +33 (4) 42 85 97 97, Fax: +33 (4) 42 85 99 66, Web portal: <https://airbusworld.helicopters.airbus.com> Technical Requests Management, or



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