Emergency Airworthiness Directive

AD No.: 2026-0119-E

Issued: 22 June 2026

Note: This Emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS S.A.S.

Type/Model designation(s):

A380 aeroplanes

Effective Date:	24 June 2026
TCDS Number(s):	EASA.A.110
Foreign AD:	Not applicable
Supersedure:	None

ATA 57 – Wings – Wing Mid Spars – Inspection

Manufacturer(s):

Airbus

Applicability:

A380-841, A380-842 and A380-861 aeroplanes, manufacturer serial numbers (MSN) 30, 42, 55, 56, 105, 142, 184, 187, 190, 202, 203, 208, 209, 227, 228 and 234.

Definitions:

For the purpose of this AD, the following definitions apply:

Groups: Group 1 are aeroplanes MSN 190, 202, 203, 209 and 228.
Group 2 are aeroplanes MSN 30, 42, 55, 56, 105, 142, 184, 187, 208, 227 and 234.

Reason:

EASA issued several ADs requiring inspection of wing spars, last one being EASA AD 2025-0280.

Following the review of the results of those inspections, it has been determined that the cracks found on certain aeroplanes could reduce the structural integrity of the wing.

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To address this potential unsafe condition, Airbus determined that an additional special detailed inspection has to be accomplished.

For the reason described above, this AD requires to contact Airbus for inspection instructions, accomplishment of those inspections and reporting of inspection results.

This AD is considered an interim action and further AD action may follow.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Inspection(s):

(1) Within the compliance time as identified in Table 1 of this AD, as applicable, contact Airbus for instructions for special detailed inspection of the Wing Mid Spars and accomplish those instructions accordingly.

Table 1 – Compliance time

Groups	Compliance Time
Group 1	Before next flight after the effective date of this AD
Group 2	Within 25 Flight Cycles (FC) after the effective date of this AD

(2) For Group 1 aeroplanes: A ferry flight (up to 3 FC, non-ETOPS, no passengers) is permitted to position the aeroplane to a location where the inspection can be accomplished.

Corrective Action(s):

(3) If, during the inspection as required by paragraph (1) of this AD, any discrepancy is detected, before next flight, contact Airbus for applicable repair instructions and, within the compliance time specified in those instructions, accomplish those instructions accordingly.

Reporting:

(4) Within 7 days after the inspection as required by paragraph (1) of this AD, or after the effective date of this AD, whichever occurs later, report the inspection results, including no findings, to Airbus.

Ref. Publications:

None.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.

2. The results of the safety assessment have indicated the need for immediate publication and notification, without the full consultation process.

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3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS SAS – EIANA (Airworthiness Office), Telephone: +33 562 110 253, Fax: +33 562 110 307, or E-mail: account.airworth-A380@airbus.com.

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