


EASA	NOTIFICATION OF A PROPOSAL TO ISSUE AN AIRWORTHINESS DIRECTIVE
	<p>PAD No.: 15-113</p> <p>Date : 01 September 2015</p> <p>Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.</p>
<p>In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below. All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation closing date indicated.</p>	
Design Approval Holder's Name: DASSAULT AVIATION	Type/Model designation(s): F7X aeroplanes
TCDS Number : EASA.A.155	
Foreign AD : Not applicable	
Supersedure : None	
ATA 53	Fuselage – Lateral Engine Pylons – Stiffener Attachments to Pylon Spar 41 Assembly – Inspection / Repair
Manufacturer(s):	Dassault Aviation
Applicability:	Falcon 7X aeroplanes serial numbers (S/N) 1 to 221 inclusive, except S/N 182 and S/N 220.
Reason:	<p>On the assembly line of Falcon 7X aeroplanes, defects were detected on left hand and right hand engine pylons. A quality review revealed that bores located on upper and lower stiffener joints to the web at pylon Frame 41 were improperly drilled. Fettlings of borings, for fixing diameter 4 mm and 5 mm, were found ovalized, too deep and having irregular surface qualities under the head of fixing. Dassault Aviation identified the individual aeroplanes that are potentially affected by this production deficiency.</p> <p>This condition, if not detected and corrected, would adversely affect the structural integrity of the aeroplane.</p> <p>To address this potential unsafe condition Dassault Aviation published Service Bulletin (SB) 7X-346 to provide corrective action instructions.</p> <p>For the reasons described above, this AD requires a one-time inspection and, depending on findings, repair of affected stiffener bores.</p>
Effective Date:	[TBD : 14 days after final AD issue date]

Required Action(s) and Compliance Time(s):	<p>Required as indicated, unless accomplished previously:</p> <p>Before exceeding 4 000 flight cycles or 98 months, whichever occurs first since first flight of the aeroplane, inspect and, depending on findings, repair bores on stiffeners at engine pylons frame 41, in accordance with the accomplishment instructions of Dassault Aviation SB 7X-346.</p>
Ref. Publications:	<p>Dassault Aviation SB 7X-346 original issue dated 06 June 2014.</p> <p>The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.</p>
Remarks :	<ol style="list-style-type: none"> 1. This Proposed AD will be closed for consultation on 29 September 2015. 2. Enquiries regarding this PAD should be referred to the Safety Information Section, Certification Directorate, EASA. E-mail ADs@easa.europa.eu. 3. For any question concerning the technical content of the requirements in this AD, please contact your Dassault Falcon Technical Assistance: <ul style="list-style-type: none"> • For Europe, Middle East and Africa based operators: Hot Line: (33) 1 47 11 37 37 • For USA, Canada and Mexico based operators: Help Desk: (1) 800-2FALCON (2325266) • All other areas: Help Desk: (1) 201 541 4747