



Airworthiness Directive

AD No.: 2016-0069

Issued: 11 April 2016

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) 216/2008 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EC) 216/2008, Article 14(4) exemption].

Design Approval Holder's Name:

AIRBUS

Type/Model designation(s):

A318, A319, A320 and A321 aeroplanes

Effective Date:

Prompted by these events, new FCD front latch and keeper assembly were developed, having a specific key necessary to un-latch the FCD. This key cannot be removed unless the FCD front latch is safely closed. The key, after removal, must be stowed in the flight deck at a specific location, as instructed in the applicable Aircraft Maintenance Manual. Applicable Flight Crew Operating Manual has been amended accordingly. After modification, the FCD is identified with a different Part Number (P/N).

For the reasons described above, this AD requires modification and re-identification of FCD.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

- (1) Within 35 months after the effective date of this AD, accomplish concurrently the actions as required by paragraphs (1.1), (1.2) and (1.3) of this AD, in accordance with the instructions of Airbus Service Bulletin (SB) A320-71-1068.
 - (1.1) Modify the left hand and right hand FCD on Engines 1 and 2.
 - (1.2) Install a placard on the box located at the bottom of the 120 VU panel, or at the bottom of the coat stowage, as applicable to aeroplane configuration.
 - (1.3) Re-identify both FCD with the new P/N, as applicable, as specified in Table 1 of this AD.

Table 1 – Fan Cowl Door P/N Change

Door Position	Old P/N	New P/N
Left Hand Side – CFM56-5A	238-0301-501	238M0301-501
	238-0	

Table 1 – Fan Cowl Door P/N Change (continued)

Door Position	Old P/N	New P/N
Right Hand Side – CFM56-5A	238-0302-501	238M0302-501
	238-0302-503	238M0302-503
	238-0302-505	238M0302-505
	238-0302-509	238M0302-509
	238-0302-511	238M0302-511
	238-0302-513	238M0302-513
	238-0302-515	238M0302-515
	238-0302-517	238M0302-517
	238-0302-519	238M0302-519
	238-0302-521	238M0302-521
	238-0302-523	238M0302-523
	238-0302-525	238M0302-525
	238-0302-527	238M0302-527
	238-0302-529	238M0302-529</

- (4) An aeroplane on which Airbus modification 157519 or modification 157521 has been embodied in production is compliant with the requirements of paragraph (1.2) of this AD.
- (5) Do not install on any aeroplane a FCD, having a P/N identified as “old” in Table 1 of this AD, as required by paragraph (5.1) or (5.2) of this AD, as applicable.
 - (5.1) For an aeroplane with a FCD installed having a P/N identified as “old” in Table 1 of this AD: After modification of that aeroplane as required by paragraph (1) of this AD.
 - (5.2) For an aeroplane that does not have a FCD installed having a P/N identified as “old” in Table 1 of this AD: From the effective date of this AD.
- (6) Installation on an aeroplane of right hand and left hand FCD, having a P/N approved after the effective date of this AD, is equal to compliance with the requirements of paragraphs (1.1) and (1.3) of this AD for that aeroplane, provided the conditions as specified in paragraphs (6.1) and (6.2) of this AD are met.
 - (6.1) The P/N must be approved by EASA, or approved under Airbus Design Organisation Approval (DOA); and
 - (6.2) The installation must be accomplished in accordance with aeroplane modification instructions approved by EASA, or approved under Airbus DOA.

Ref. Publications:

Airbus SB A320-71-1068 original issue dated 18 December 2015.

The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 09 February 2016 as PAD 16-022 for