



Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 18-033

Issued: 07 March 2018

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EC) 216/2008 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name:

ROLLS-ROYCE plc

Type/Model designation(s):

Trent 1000 engines

Effective Date: [TBD – standard: 14 days after Final AD issue date]

TCDS Number(s): EASA.E.036

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2017-0017R2 dated 15 September 2017.

ATA 72 – Engine – Intermediate Pressure Compressor Rotor Seal – Inspection / De-Pairing Limitations

Manufacturer(s):

Rolls-Royce plc (RR)

Applicability:

Trent 1000-A2, Trent 1000-C2, Trent 1000-D2, Trent 1000-E2, Trent 1000-G2, Trent 1000-H2, Trent 1000-J2, Trent 1000-K2, Trent 1000-L2, Trent 1000-AE2 and Trent 1000-CE2 engines, all serial numbers (ESN).

These engines are known to be installed on, but not limited to, Boeing 787-8 and 787-9 aeroplanes.

Definitions:

For the purpose of this AD, the following definitions apply:

The NMSB: RR Alert Non-Modification Service Bulletin (NMSB) TRENT 1000 72-AJ467 Revision 3. The NMSB has an 'A' (Alert) in the number, but a later revision may not have that 'A'. This kind of change does not effectively alter the publication references.

Affected seal: Intermediate pressure compressor (IPC) rotor seals, Part Number (P/N) KH19098.



Groups: Group 1 engines are those that have an affected seal installed. Group 2 engines are those that do not have an affected seal installed. An engine in post-modification (mod) 72-J603 or post-mod 72-J704 configuration, ESN 10554 and higher, is a Group 2 engine, provided the engine remains in that configuration.

Reason:

A low speed abort (60 to 65 knots) occurred on take-off on a Trent 1000-powered Boeing 787 aeroplane. The pilot performed a commanded engine shutdown and the aeroplane safely returned to the gate. Following investigation, failure and release of the IPC rotor seal was confirmed as having caused this event. RR have confirmed that other IPC rotor seals, P/N KH19098, have been found with cracking at the seal head.

This condition, if not detected and corrected, could lead to engine power loss, possibly resulting in reduced control of the aeroplane.

To address this potential unsafe condition, RR published the NMSB, providing inspection instructions. Consequently, EASA issued AD 2017-0017 (later revised) to require repetitive borescope inspections of the affected IPC rotor seals and, depending on findings, accomplishment of applicable corrective action(s). Post-mod/SB 72-J603 engines, having a revised IPC buffer static seal assembly, and post-mod/SB 72-J704 engines, having a revised IPC rotor buffer seal, are not affected. RR published SB TRENT 1000 72-J603 and SB TRENT 1000 72-J704 to provide modification instructions for in-service engines. RR also published Revision 2 of the NMSB, introducing a conditional terminating action.

Since EASA AD 2017-0017R2 was issued, RR revised Alert NMSB TRENT 1000 72-AJ467 (the NMSB) and issued Alert NMSB TRENT 1000 72-AJ929, which is the subject of [EASA PAD 18-034](#), introducing new de-pairing instructions for engines in post-mod/SB 72-J704 configuration, which affects the de-pairing requirements of EASA AD 2017-0017R2.

For the reasons described above, this AD retains the requirements of EASA AD 2017-0017R2, which is superseded, and amends the de-pairing requirements/limitations accordingly.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Repetitive Inspections:

- (1) For Group 1 engines: Within the compliance times as specified in the NMSB, accomplish an on-wing borescope inspection of the affected seal in accordance with the instructions of Section 3, Part A (front face) of the NMSB, and/or, depending on findings (see Figure 1 of the NMSB), Part B (rear face) of the NMSB, as applicable. Thereafter, depending on findings, repeat the on-wing borescope inspection at intervals not to exceed the value(s) as specified in Figures 1, 2 or 4 of the NMSB, as applicable.
- (2) An in-shop inspection in accordance with the instructions of RR NMSB TRENT 1000 72-J353 may be substituted for an on-wing inspection as required by paragraph (1) of this AD, provided the compliance times are not exceeded.



Limitations:

- (3) From 15 February 2017 [the effective date of the original issue of EASA AD 2017-0017], do not operate an aeroplane, having two Group 1 engines installed that are both subject to the 20 engine flight cycles (EFC) inspection interval (see Figure 4 of the NMSB) as required by paragraph (1) of this AD.
- (4) From the effective date of this AD, do not operate an aeroplane, having a Group 1 engine installed that is subject to 20 EFC interval inspections (see Figure 4 of the NMSB) as required by paragraph (1) of this AD; and a Group 2 engine installed that is subject to 50 EFC interval inspections, or to a single 100 EFC fly-on period, as specified in Figures 1 and 2 of RR Alert NMSB TRENT 1000 72-AJ929 [see EASA PAD 18-034].

Corrective Action(s):

- (5) If, during any on-wing inspection as required by paragraph (1) of this AD, or any in-shop inspection as specified in paragraph (2) of this AD, as applicable, any crack is found on the rear face of the affected seal, that is at or beyond the reject limits as specified in the NMSB, before next flight, or before release to service of the engine, as applicable, replace the affected seal with a new part.

Credit:

- (6) Inspections and corrective actions on an engine, accomplished before 15 February 2017 [the effective date of the original issue of EASA AD 2017-0017] in accordance with the instructions of the NMSB (on wing) at original issue, or Revision 1, or Revision 2, or that meet the intent of the NMSB, or RR NMSB TRENT 1000 72-J353 (in-shop), are acceptable to comply with the initial requirements of paragraphs (1) and (5) of this AD, for that engine.

Terminating Action(s):

- (7) Modification of an engine in accordance with the instructions of RR SB TRENT 1000 72-J603 or SB TRENT 1000 72-J704 constitutes terminating action for the repetitive inspections as required by paragraph (1) of this AD for that engine.
- (8) For a Group 1 engine on which an affected seal, by the results of two consecutive inspections of the front face in accordance with the instructions of Section 3, Part A, of the NMSB, is confirmed to be without cracks, no further inspections are required by this AD on that engine, except as required by paragraph (10) of this AD.

Part(s) Installation:

- (9) From 15 February 2017 [the effective date of the original issue of this AD], it is allowed to install an affected seal on any Group 1 or Group 2 engine, provided that, following installation, inspections on that engine are (re-)started as required by paragraph (10) of this AD.

Re-starting Inspections:

- (10) Before release to service of an engine after maintenance action on the O2 module, where a new P/N KH19094 IPC static seal and an affected seal are installed, or after replacement of the O2 module, as applicable, contact RR to determine when the repetitive inspections, as required by paragraph (1) of this AD, must be re-started on that engine.



Ref. Publications:

Rolls-Royce NMSB TRENT 1000 72-J353, original issue, dated 25 August 2016.

Rolls-Royce SB TRENT 1000 72-J603, original issue, dated 12 October 2016.

Rolls-Royce SB TRENT 1000 72-J704, original issue, dated 23 June 2017.

Rolls-Royce Alert NMSB TRENT 1000 72-AJ467, original issue, dated 9 November 2016, or Revision 1 dated 13 February 2017, or Revision 2 dated 2 June 2017, or Revision 3 dated 14 February 2018.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 04 April 2018.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
3. For any question concerning the technical content of the requirements in this AD, please contact your designated Rolls-Royce representative, or download the publication from your Rolls Royce Care account at <https://customers.rolls-royce.com>.

If you do not have a designated representative or Rolls-Royce Care account, please contact **Corporate Communications** at **Rolls-Royce plc**, P.O. Box 31, Derby, DE24 8BJ, United Kingdom Telephone +44 (0)1332 242424, or

send an email through http://www.rolls-royce.com/contact/civil_team.jsp identifying the correspondence as being related to **Airworthiness Directives**.

