



# Notification of a Proposal to issue an Airworthiness Directive

**PAD No.:** 20-059

**Issued:** 27 March 2020

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

**Design Approval Holder's Name:**

PILATUS AIRCRAFT Ltd

**Type/Model designation(s):**

PC-24 aeroplanes

**Effective Date:** [TBD - standard: 14 days after AD issue date]

**TCDS Number(s):** EASA.A.594

**Foreign AD:** Not applicable

**Supersedure:** None

## ATA – Airplane Flight Manual – Section Normal Procedures / Electronic Circuit Breaker – Amendment

**Manufacturer(s):**

Pilatus Aircraft Ltd

**Applicability:**

PC-24 aeroplanes, all manufacturers serial numbers (MSN).

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The AFM TR:** Pilatus Aircraft PC-24 Airplane Flight Manual (AFM) Temporary Revision (TR) 02371-016 ("Change of ECB Check In Before Engine Start Procedure").

**Reason:**

An occurrence was reported where, before take-off after maintenance of a PC-24 aeroplane, some electronic circuit breakers (ECB) were found in a "LOCKED" state.

This condition, if not corrected, could lead to a loss of power supply to equipment, without indication to the flight crew before take-off.



To address this potential unsafe condition, Pilatus issued the AFM TR, as defined in this AD, to provide operators with the necessary preflight check instructions.

For the reason described above, this AD requires amendment of the AFM.

#### **Required Action(s) and Compliance Time(s):**

Required as indicated, unless accomplished previously:

#### **AFM Amendment:**

- (1) Within 30 days after the effective date of this AD, amend the AFM by inserting a copy of the AFM TR, inform all flight crews and, thereafter, operate the aeroplane accordingly.
- (2) Amending the AFM to incorporate a later revision, which includes the AFM change as required in this AD, is acceptable to comply with the requirements of paragraph (1) of this AD.

#### **Ref. Publications:**

Pilatus PC-24 AFM TR 02371-016 original issue dated 01 November 2019.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

#### **Remarks:**

1. This Proposed AD will be closed for consultation on 24 April 2020.
2. Enquiries regarding this PAD should be referred to the EASA Programming and Continued Airworthiness Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the [EU aviation safety reporting system](#).
4. For any question concerning the technical content of the requirements in this PAD, please contact: Pilatus Aircraft Ltd, Customer Support General Aviation, CH-6371 Stans, Switzerland Telephone: +41 848 24 7 365, E-mail: [techsupport.ch@pilatus-aircraft.com](mailto:techsupport.ch@pilatus-aircraft.com), Website: [www.pilatus-aircraft.com](http://www.pilatus-aircraft.com).

