



Airworthiness Directive

AD No.: 2025-0069

Issued: 28 March 2025

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

ROLLS-ROYCE DEUTSCHLAND Ltd & Co KG

Type/Model designation(s):

RB211 Trent 900 engines

Effective Date: 11 April 2025

TCDS Number(s): EASA.E.012

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2024-0039 dated 08 February 2024.

ATA 05 – Time Limits / Maintenance Checks – Engine Time Limits Manual – Amendment

Manufacturer(s):

Rolls-Royce plc

Applicability:

RB211 Trent 970-84, 970B-84, 972-84, 972B-84, 972E-84, 977-84, 977B-84 and 980-84 engines, all serial numbers.

These engines are known to be installed on, but not limited to, Airbus A380 aeroplanes.

Definitions:

For the purpose of this AD, the following definitions apply:

The ALS: The airworthiness limitation section (ALS) for Rolls-Royce RB211 Trent 900 engines, as published in the Time Limits Manual (TLM) T-TRENT-9RR Revision 68, module 05-10-01-800-801 (Critical Group A Parts Lives) and module 05-20-01-800-801 (Mandatory Inspections).

The AMP: The Aircraft Maintenance Programme (AMP) contains the tasks on the basis of which the scheduled maintenance is conducted to ensure the continuing airworthiness of each operated aircraft. For engines installed on aircraft operated under EU regulations, the operator or the owner ensures compliance with the AMP as stipulated in Commission Regulation (EU) [1321/2014](#).



New and/or more restrictive tasks: This includes all tasks that are new and all tasks for which a threshold or interval was reduced, which were introduced into the ALS (as defined in this AD) since the previous ALS revision that is currently incorporated in the AMP.

Reason:

The airworthiness limitations for the RB211 Trent 900 engines, which are approved by EASA, are currently defined and published in the ALS. These instructions have been identified as mandatory for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.

Previously, EASA issued AD 2024-0039 to require accomplishment of the tasks and limitations specified in TLM T-TRENT-9RR Revision 65 dated 01 October 2023.

Since that AD was issued, Rolls-Royce issued the ALS, as defined in this AD, adding new and more restrictive tasks. Additionally, Rolls-Royce issued Technical Variation (TV) 257089, which is an advanced issue of Rolls-Royce RB211 Trent 900 Cleaning, Inspection and Repair (CIR) Manual Task 72-00-00-200-801, referenced in TLM Task 05-20-01-800-801, and it incorporates equivalent inspections to those published in Rolls-Royce RB211 Trent 900 CIR Manual at Revision 181.

For the reasons described above, this AD supersedes EASA AD 2024-0039, requires accomplishment of the actions specified in the ALS and includes credit for inspections accomplished in accordance with the Rolls-Royce TV 257089.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Maintenance Tasks and Replacement of Life Limited Parts:

- (1) From the effective date of this AD, accomplish the following actions, as specified in the ALS, as defined in this AD, as applicable to engine model and depending on engine configuration:
 - (1.1) Replace each component before exceeding the applicable life limit, and
 - (1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.

Corrective Action(s):

- (2) In case of finding discrepancies (as defined in the ALS) during accomplishment of any task as required by paragraph (1) of this AD, within the compliance time specified in the ALS, accomplish the applicable corrective action(s) in accordance with the approved Rolls-Royce maintenance documentation. If no compliance time is identified in the ALS, accomplish the applicable corrective action(s) before next flight. If a detected discrepancy is not identified in the ALS, before next flight, contact Rolls-Royce for approved instructions and accomplish those instructions accordingly.



AMP Revision:

- (3) Within 12 months after the effective date of this AD, revise the approved AMP by incorporating the limitations, tasks and associated thresholds and intervals described in the ALS, as applicable to engine model and depending on engine configuration.

Credit:

- (4) If, before the effective date of this AD, the AMP has been revised to incorporate the maintenance tasks and life limitations as specified in a previous revision of Rolls-Royce Trent 900 TLM T-TRENT-9RR, module 05-10-01-800-801 and module 05-20-01-800-801, that action ensures the continued accomplishment of those tasks and limitations.

Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations, as applicable to engine model and depending on engine configuration, as defined in, and within the compliance times as specified in the ALS, to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and/or more restrictive tasks and limitations as defined in the ALS, as applicable to engine model and depending on engine configuration, into the AMP to comply with paragraph (3) of this AD.

- (5) Inspections of the IP8 Air Tubes and, depending on findings, corrective actions, accomplished on an engine before the effective date of this AD in accordance with the instructions of the Rolls-Royce TV 257089 is an acceptable method to comply with the TLM Task 05-20-01-800-801 Section R, referencing Rolls-Royce RB211 Trent 900 CIR Manual Task 72-00-00-200-801, for that engine. After the effective date of this AD, the inspections shall be accomplished in accordance with the instructions of the TLM Task 05-20-01-800-801.

Recording AD Compliance:

- (6) When the AMP of an aeroplane has been revised as required by paragraph (3) or (4) of this AD, as applicable, that action ensures continued accomplishment of the tasks as required by paragraphs (1) and (2) of this AD for an affected engine installed on that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.



Ref. Publications:

Rolls-Royce Trent 900 TLM T-TRENT-9RR Revision 68 dated 24 February 2025.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 13 November 2024 as PAD 24-136 for consultation until 11 December 2024 and republished on 10 March 2025 as PAD 24-136R1 for additional consultation until 24 March 2025. The Comment Response Documents can be found in the [EASA Safety Publications Tool](#), in the compressed (zipped) file attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact your designated Rolls-Royce representative, or download the publication from your Rolls-Royce Care account at <https://customers.rolls-royce.com>.

If you do not have a designated representative or Rolls-Royce Care account, please contact **Corporate Communications** at **Rolls-Royce plc**, P.O. Box 31, Derby, DE24 8BJ, United Kingdom Telephone +44 (0)1332 242424,

or send an email through <https://www.rolls-royce.com/contact-us/civil-aerospace.aspx> identifying the correspondence as being related to **Airworthiness Directives**.

