

Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 24-136R1

Issued: 10 March 2025

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name: Type/Model designation(s):

ROLLS-ROYCE DEUTSCHLAND Ltd & Co KG RB211 Trent 900 engines

Effective Date: [TBD - standard: 14 days after AD issue date]

TCDS Number(s): EASA.E.012

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2024-0039 dated 08 February 2024.

ATA 05 – Time Limits / Maintenance Checks – Engine Time Limits Manual – Amendment

Manufacturer(s):

Rolls-Royce plc

Applicability:

RB211 Trent 970-84, 970B-84, 972-84, 972B-84, 972E-84, 977-84, 977B-84 and 980-84 engines, all serial numbers.

These engines are known to be installed on, but not limited to, Airbus A380 aeroplanes.

Definitions:

For the purpose of this AD, the following definitions apply:

The ALS: The airworthiness limitation section (ALS) for Rolls-Royce RB211 Trent 900 engines, as published in the Time Limits Manual (TLM) T-TRENT-9RR Revision 68, module 05-10-01-800-801 (Critical Group A Parts Lives) and module 05-20-01-800-801 (Mandatory Inspections).

The AMP: The Aircraft Maintenance Programme (AMP) contains the tasks on the basis of which the scheduled maintenance is conducted to ensure the continuing airworthiness of each operated



aircraft. For engines installed on aircraft operated under EU regulations, the operator or the owner ensures compliance with the AMP as stipulated in Commission Regulation (EU) <u>1321/2014</u>.

New and/or more restrictive tasks: This includes all tasks that are new and all tasks for which a threshold or interval was reduced, which were introduced into the ALS (as defined in this AD) since the previous ALS Revision that is currently incorporated in the AMP.

Reason:

The airworthiness limitations for the RB211 Trent 900 engines, which are approved by EASA, are currently defined and published in the ALS. These instructions have been identified as mandatory for continued airworthiness.

Failure to accomplish these instructions could result in an unsafe condition.

Previously, EASA issued AD 2024-0039 to require accomplishment of the tasks and limitations specified in TLM T-TRENT-9RR Revision 65 dated 01 October 2023.

Since that AD was issued, Rolls-Royce issued the ALS, as defined in this AD, adding new and more restrictive tasks.

For the reasons described above, this AD supersedes EASA AD 2024-0039 and requires accomplishment of the actions specified in the ALS.

Since the original issue of this PAD was issued, comments were received indicating the need for update of Rolls-Royce RB211 Trent 900 TLM T-TRENT-9RR at Revision 67, which was originally referenced in the EASA PAD 24-136; consequently, Rolls-Royce issued Trent 900 TLM T-TRENT-9RR at Revision 68. Additionally, it was acknowledged that Rolls-Royce Technical Variation (TV) 257089 is an advanced issue of Rolls-Royce RB211 Trent 900 Cleaning, Inspection and Repair (CIR) Manual Task 72-00-00-200-801, which is referenced in TLM Task 05-20-01-800-801, and it incorporates equivalent inspections to those published in Rolls-Royce RB211 Trent 900 CIR Manual at Revision 181.

This PAD is republished for additional consultation adding reference to the Rolls-Royce RB211 Trent 900 TLM T-TRENT-9RR at Revision 68 and including credit for inspections accomplished in accordance with the Rolls-Royce TV 257089.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Maintenance Tasks and Replacement of Life Limited Parts:

- (1) From the effective date of this AD, accomplish the following actions, as specified in the ALS, as defined in this AD, as applicable to engine model and depending on engine configuration:
 - (1.1) Replace each component before exceeding the applicable life limit, and
 - (1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.



Corrective Action(s):

(2) In case of finding discrepancies (as defined in the ALS) during accomplishment of any task as required by paragraph (1) of this AD, within the compliance time specified in the ALS, accomplish the applicable corrective action(s) in accordance with the approved Rolls-Royce maintenance documentation. If no compliance time is identified in the ALS, accomplish the applicable corrective action(s) before next flight. If a detected discrepancy is not identified in the ALS, before next flight, contact Rolls-Royce for approved instructions and accomplish those instructions accordingly.

AMP Revision:

(3) Within 12 months after the effective date of this AD, revise the approved AMP by incorporating the limitations, tasks and associated thresholds and intervals described in the ALS, as applicable to engine model and depending on engine configuration.

Credit:

(4) If, before the effective date of this AD, the AMP has been revised to incorporate the maintenance tasks and life limitations as specified in a previous revision of Rolls-Royce Trent 900 TLM T-TRENT-9RR, module 05-10-01-800-801 and module 05-20-01-800-801, that action ensures the continued accomplishment of those tasks and limitations.

Consequently, for an aeroplane to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations, as applicable to engine model and depending on engine configuration, as defined in, and within the compliance times as specified in the ALS, to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and/or more restrictive tasks and limitations as defined in the ALS, as applicable to engine model and depending on engine configuration, into the AMP to comply with paragraph (3) of this AD.

(5) Inspections of the IP8 Air Tubes and, depending on findings, corrective actions, accomplished on an engine before the effective date of this AD in accordance with the instructions of the Rolls-Royce TV 257089 is an acceptable method to comply with the TLM Task 05-20-01-800-801 Section R, referencing Rolls-Royce RB211 Trent 900 CIR Manual Task 72-00-00-200-801, for that engine. After the effective date of this AD, the inspections shall be accomplished in accordance with the instructions of the TLM Task 05-20-01-800-801.

Recording AD Compliance:

(6) When the AMP of an aeroplane has been revised as required by paragraph (3) or (4) of this AD, as applicable, that action ensures continued accomplishment of the tasks as required by paragraphs (1) and (2) of this AD for an affected engine installed on that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, as applicable, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.



Ref. Publications:

Rolls-Royce Trent 900 TLM T-TRENT-9RR Revision 68 dated 24 February 2025.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

- 1. This Proposed AD will be closed for consultation on 24 March 2025.
- 2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
- 3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the <u>EU aviation safety reporting system</u>. This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
- 4. For any question concerning the technical content of the requirements in this PAD, please contact your designated Rolls-Royce representative, or download the publication from your Rolls-Royce Care account at https://customers.rolls-royce.com.

If you do not have a designated representative or Rolls-Royce Care account, please contact **Corporate Communications** at **Rolls-Royce plc**, P.O. Box 31, Derby, DE24 8BJ, United Kingdom Telephone +44 (0)1332 242424,

or send an email through https://www.rolls-royce.com/contact-us/civil-aerospace.aspx identifying the correspondence as being related to **Airworthiness Directives**.

