

## Airworthiness Directive

**AD No.:** 2025-0219

**Issued:** 06 October 2025

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

**Design Approval Holder's Name:**

AIRBUS S.A.S.

**Type/Model designation(s):**

A330-743L aeroplanes

**Effective Date:** 20 October 2025

**TCDS Number(s):** EASA.A.004

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA 05 – Time Limits / Maintenance Checks – Airworthiness Limitations Section Part 1 – Safe Life Airworthiness Limitation Items – Amendment

**Manufacturer(s):**

Airbus

**Applicability:**

Airbus A330-743L aeroplanes, all manufacturer serial numbers.

**Definitions:**

For the purpose of this AD, the following definitions apply:

**The Variation:** Airbus A330-700L Airworthiness Limitations Section (ALS) Part 1 Variation 0.4.

**The AMP:** The Aircraft Maintenance Programme (AMP) contains the tasks on the basis of which the scheduled maintenance is conducted to ensure the continuing airworthiness of each operated aeroplane. For aeroplanes operated under EU regulation the operator or the owner ensures compliance with the AMP as stipulated in Commission Regulation (EU) [1321/2014](#).

**New and/or more restrictive tasks:** This includes all tasks that are new and all tasks for which a threshold or interval was reduced, which were introduced into the ALS (as defined in this AD) since the previous ALS Revision that is currently incorporated in the AMP.

**Reason:**

The airworthiness limitations instructions for the Airbus A330-743L aeroplanes, which are approved by EASA, are currently defined and published in the A330-700 ALS document. These instructions have been identified as mandatory for continued airworthiness. The airworthiness limitations applicable to the Safe Life Airworthiness Limitations Items are specified in ALS Part 1.

Failure to accomplish these instructions could result in an unsafe condition.

EASA previously issued AD 2025-0049, requiring the accomplishment of the actions described in Variation 0.1, Variation 0.2 and Variation 0.3 of the ALS Part 1.

Since that AD was issued, new and/or more restrictive tasks have been developed, and introduced in the Variation, as defined in this AD.

For the reason described above, this AD requires accomplishment of the actions specified in the Variation.

EASA AD 2025-0049 is not superseded by this AD.

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the actions required by this AD have been already accomplished:

**Replacement of Life Limited Parts:**

- (1) From the effective date of this AD, replace each component before exceeding the applicable life limit, as specified in the Variation (see Note 1 of this AD), as applicable, depending on aeroplane configuration.

Where this AD requires a task which is already required per EASA AD 2025-0049, the instructions of the Variation invalidate the instructions of Variation 0.1, Variation 0.2 and Variation 0.3 of the ALS Part 1, as applicable.

Note 1: For the purpose of this AD, the thresholds, as defined in the 'Compliance Time' pages of the Variation, include specific compliance times for certain tasks.

**AMP Revision:**

- (2) Within 12 months after the effective date of this AD, revise the AMP by incorporating the tasks and associated thresholds and intervals described in the Variation, as applicable, depending on aeroplane configuration.

**Recording AD Compliance:**

- (3) When the AMP of an aeroplane has been revised as required by paragraph (2) of this AD, as applicable, that action ensures continued accomplishment of the tasks as required by paragraphs (1) and (2) of this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) of this AD, it is not necessary that accomplishment of individual action is recorded for demonstration of AD compliance on a continued basis.



**Ref. Publications:**

Airbus A330-700L ALS Part 1 Variation 0.4 dated 09 July 2025.

The use of later approved variations or revisions of the above-mentioned documents, or of an ALS revision which includes the content of the above-mentioned documents, is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 08 August 2025 as PAD 25-125 for consultation until 05 September 2025. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – 1IAL (Airworthiness Office), E-mail: [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com).

