



Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 26-018

Issued: 29 January 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name:

SAFRAN ELECTRONICS & DEFENSE
BEACONS

Type/Model designation(s):

Emergency Locator Transmitter KANNAD
406 COMPACT

Effective Date: [TBD - standard: 14 days after AD issue date]

ETSO Authorisation: EASA.210.818 Rev. B

Foreign AD: Not applicable

Supersedure: None

ATA 25 – Equipment / Furnishings – Emergency Locator Transmitter – Modification

Manufacturer(s):

Safran Electronics & Defense Beacons SAS, formerly Orolia SAS, Kannad and Martec Serpe-lesm

Applicability:

Kannad 406 AF-Compact Emergency Locator Transmitter (ELT) model, having Part Number (P/N) S1840501-01 or P/N S1840501-04, all serial numbers.

The ELT may be eligible for installation on any aircraft and may have been installed during the aircraft manufacturing process (production line), or in-service modification, either through a Supplemental Type Certificate, or using Type Certificate holder approved modification instructions, or through a minor modification approval.

Definitions:

For the purpose of this AD, the following definitions apply:

The SB: Safran Electronics & Defense Beacons Service Bulletin (SB) S1840501-25-01.



Affected part: Programming dongle connector having P/N S1820514-01 or a Programming Dongle Assembly having P/N S1820514-06.

Serviceable part: DIN 12 connector having P/N S1820514-03.

Groups: Group 1 are those ELT which installation configuration includes an affected part. Group 2 are those which are not Group 1.

Reason:

Occurrences have been reported of Kannad 406 AF-Compact ELT transmitting random bursts in maintenance code during valid distress transmissions. Relevant investigations determined that certain ELT configurations may be affected.

This condition, if not corrected, could lead to erroneously encoded emissions during an aircraft emergency, possibly delaying the arrival of the rescue services and timely medical assistance to injured crew members or passengers.

To address this potential unsafe condition, Safran Electronics & Defense Beacons issued the SB, providing instructions to modify the installation configuration.

For the reason described above, this AD requires modification of ELT installation. This AD also regulates (re)installation of affected parts.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Modification:

- (1) For Group 1: Within 12 months after the effective date of this AD, modify the ELT installation (replacing the affected part with a serviceable part, or removing the affected part, as applicable) in accordance with the instructions of the SB.

Part(s) Installation:

- (2) Do not install an affected part on any ELT as required by paragraph 2.1 or 2.2 of this AD, as applicable (see Note 1 of this AD):

(2.1) For Group 1: After the modification as required by paragraph (1) of this AD.

(2.2) For Group 2: From the effective date of this AD.

Note 1: Temporary installation of an affected part on the ELT, accomplished during maintenance activities, is not considered as 'install' as specified in paragraph (2) of this AD.

Ref. Publications:

Safran Electronics & Defense Beacons SB S1840501-25-01 Revision 00 (original issue) dated 21 January 2026.



The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 26 February 2026.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the [EU Aviation Safety Reporting System](#). This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA parts manufacturer approval (PMA), Supplemental Type Certificate (STC) or other modification.
4. For any question concerning the technical content of the requirements in this PAD, please contact: SAFRAN Electronics & Defense Beacons - ZI des Cinq Chemins, 56520, Guidel, France; Telephone: +33 (0) 2 97 02 49 00; Email: support.sar@beacons.safrangroup.com.

