



# Notification of a Proposal to issue an Airworthiness Directive

**PAD No.:** 26-021

**Issued:** 02 February 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

**Design Approval Holder's Name:**

AIRBUS DEFENCE AND SPACE S.A.

**Type/Model designation(s):**

C-295 aeroplanes

**Effective Date:** [TBD - standard: 14 days after AD issue date]

**TCDS Number(s):** EASA.A.186

**Foreign AD:** Not applicable

**Supersedure:** None

## ATA 92 – Electrical System Installation – Inspections

### Manufacturer(s):

Airbus Defence and Space (DS), S.A.U., formerly EADS Construcciones Aeronáuticas, S.A.U. (EADS-CASA), Construcciones Aeronáuticas S.A. (CASA)

### Applicability:

C-295 aeroplanes, manufacturer serial numbers (MSN) 188, 189 and MSN 191.

### Definitions:

For the purpose of this AD, the following definitions apply:

**The AOT:** Airbus DS Alert Operators Transmission (AOT) AOT-C295-92-0001.

**Affected area:** Area adjacent to frame (FR) 30, between stringer (STR) 6 and STR 7 on right-hand side, as identified in the AOT.

**Affected part:** Harness protection adjacent to affected area.



**Serviceable part:** Harness protection, eligible for installation in the affected area in accordance with Airbus DS instructions, which is new, or which has passed the inspection (no finding detected) in accordance with the instructions of the AOT.

**Reason:**

It was determined that inspection of the affected part which is required to comply with the applicable Electrical Wiring Interconnection System certification requirements, is not included in the published Airworthiness Limitations Section (ALS) for certain aeroplanes with a specific modification configuration, commercially designated as C-295FW.

Failure to accomplish the inspection of the affected part could lead to failures of the aeroplane systems connected to, or dependent on, the affected wires and/or fires in the cabin.

For the reason described above, this AD requires repetitive Detailed Visual Inspections (DVI) of the affected part, as defined in this AD, and, depending on findings, replacement of the affected part.

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

**Inspection(s):**

- (1) Within 50 flight hours (FH) after the effective date of this AD and, thereafter, at intervals not to exceed 350 FH, accomplish a DVI of the affected part in accordance with the instructions of the AOT.

**Corrective Action(s):**

- (2) If, during any inspection as required by paragraph (1) of this AD, any discrepancy is detected as defined in the AOT, before next flight replace the affected part with a serviceable part in accordance with approved Airbus DS maintenance instructions.

**Terminating Action:**

- (3) None.

**Ref. Publications:**

Airbus DS AOT-C295-92-0001 Revision 0 (original issue) dated 29 September 2025.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. This Proposed AD will be closed for consultation on 02 March 2026.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred



on a product, part or appliance not affected by this PAD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

4. For any question concerning the technical content of the requirements in this PAD, please contact: Airbus DS Services / Engineering Support, Fax: +34 91 585 3127, E-mail: [MTA.TechnicalService@airbus.com](mailto:MTA.TechnicalService@airbus.com).

For North American operators, contact alternatively:

E-mail: [TechnicalSupport@airbusmilitaryna.com](mailto:TechnicalSupport@airbusmilitaryna.com).

