



Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 26-052

Issued: 20 April 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Approval Holder's Name:

AIRBUS HELICOPTERS

Type/Model designation(s):

AS 350 and EC 130 helicopters

Effective Date: [TBD - 14 days after AD issue date]

TCDS Number(s): EASA.R.008

Foreign AD: Not applicable

Supersedure: This AD supersedes EASA AD 2006-0337 dated 07 November 2006.

ATA 80 – Starting - Starter Generator Limitation Label - Installation

Manufacturer(s):

Airbus Helicopters (AH), formerly Eurocopter, Eurocopter France, Aerospatiale

Applicability:

AS 350 B3 helicopters, all serial numbers equipped with a starter generator having part number (P/N) 200SGL130Q or P/N 200SGL130Q-4, except those in any of the following configuration:

- Equipped with a SAFRAN (formerly Turboméca) ARRIEL 2D engine.
- AH modification (mod) 073345 embodied.
- AH mod 073354 embodied.

EC 130 B4 helicopters, all serial numbers equipped with a starter generator having P/N 200SGL130Q or P/N 200SGL130Q-4, except those in any of the following configuration:

- AH mod 073345 embodied.
- AH mod 073354 embodied.

Definitions:

For the purpose of this AD, the following definitions apply:



The ASB: AH Alert Service Bulletin (ASB) AS350-01.00.57 Revision (Rev) 2 or AH ASB EC130-04A002 Rev 2, as applicable.

Groups:

Group 1 helicopters are those on which SKURKA (formerly APC) 200-ampere starter generator (P/N) 200SGL130Q is installed.

Group 2 helicopters are those on which SKURKA 200-ampere starter generator P/N 200SGL130Q-4 is installed.

The additional limitation: The maximum continuous current supplied by the starter generator must be limited to 180 Amperes (A) to prevent engine surging.

Reason:

An investigation determined that the power drawn by SKURKA 200-ampere starter generators from the engines is above the declared consumption capacity. The maximum continuous intensity that can be drawn from the starter generator must be limited to 180 A.

This condition, if not corrected, could lead to an engine surge, possibly resulting in engine failure and reduced control of the helicopter.

To address this potential unsafe condition, AH published the original issue of the ASB, as defined in this AD, providing instructions to implement the limitation and to install a placard for the maximum intensity that can be drawn from the starter generator, and EASA published AD 2006-0337 to require the accomplishment of those actions.

Since that AD was issued, AH revised the ASB (now at Rev 02), as defined in this AD, modifying its applicability to include the starter generator having P/N 200SGL130Q-4 and to exclude those helicopters equipped with a SAFRAN ARRIEL 2D engine or with certain modifications embodied.

For the reason described above, this AD retains the requirements of EASA AD 2006-0337, which is superseded, and expands the applicability.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Placard Installation:

- (1) For Group 1 helicopters: Within 110 flight hours (FH) or 12 months, whichever occurs first after 17 November 2006 [the effective date of EASA AD 2006-0337], install a placard indicating the new limitation on the instrument panel below the Vehicle Engine Multifunction Display (VEMD) in accordance with the instructions of the ASB.
- (2) For Group 2 helicopters: Within 110 FH or 12 months, whichever occurs first after the effective date of this AD, install a placard indicating the new limitation on the instrument panel below the VEMD in accordance with the instructions of the ASB.



Credit:

- (3) For Group 1 helicopters: Installation of a placard indicating the new limitation, accomplished before the effective date of this AD in accordance with the instructions of the ASB at original issue or at Rev 1 is acceptable for compliance with the requirement of paragraph (1) of this AD for that helicopter.

Operational Limitation:

- (4) For Group 1 helicopters: From 17 November 2006 [the effective date of EASA AD 2006-0337], implement the additional limitation, as defined in this AD.
- (5) For Group 2 helicopters: From the effective date of this AD, implement the additional limitation, as defined in this AD.
- (6) Amending the Rotorcraft Flight Manual (RFM) of a helicopter in accordance with the instructions of the ASB is an acceptable method to comply with the requirements of paragraphs (4) or (5), as applicable, for that helicopter.

Ref. Publications:

AH ASB EC130-04A002 Revision 0 dated 10 August 2006, Revision 1 dated 14 September 2006 or Revision 2 dated 15 April 2026.

AH ASB AS350-01.00.57 Revision 0 dated 10 August 2006, Revision 1 dated 14 September 2006 or Revision 2 dated 15 April 2026.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 04 May 2026.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
4. For any question concerning the technical content of the requirements in this PAD, please contact: Airbus Helicopters Customer Support, Telephone +33 (0)4.42.85.97.89, Fax + 33 (0)4.42.85.99.66, E-mail: Airframe.Technical-Support@airbus.com, Keycopter Technical Request Management: TechnicalSupport.Helicopters@airbus.com.

