



## Notification of a Proposal to issue an Airworthiness Directive

**PAD No.:** 26-061

**Issued:** 07 May 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

**Design Approval Holder's Name:**

AIRBUS DEFENCE AND SPACE S.A.

**Type/Model designation(s):**

C-295 aeroplanes

**Effective Date:** [TBD - standard: 14 days after AD issue date]

**TCDS Number(s):** EASA.A.186

**Foreign AD:** Not applicable

**Supersedure:** None

### ATA 34 – Navigation – Radio Altimeter and Ground Proximity System Antennas – Modification

#### Manufacturer(s):

Airbus Defence and Space (DS), S.A.U., formerly EADS Construcciones Aeronáuticas, S.A.U. (EADS-CASA), Construcciones Aeronáuticas S.A. (CASA)

#### Applicability:

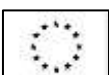
C-295 aeroplanes, manufacturer serial numbers (MSN) 188, 189 and MSN 191.

#### Definitions:

For the purpose of this AD, the following definitions apply:

**The SB1:** Airbus DS Service Bulletin (SB) SB-295-34-0091C.

**The SB2:** Airbus DS SB-295-34-0092C.



**Reason:**

It was determined that during the installation of certain RALT and GPS antenna supports, existing countersunk rivets were removed, and similar rivets were reinstalled on the doubler, leaving empty (or filled only with sealant) the space previously occupied by the removed rivets countersunk head.

This condition, if not corrected, could affect the structural integrity of the aeroplane.

To address this potential unsafe condition, Airbus DS issued the SB1 (modification of RALT antenna installation) and the SB2 (modification of GPS antenna installation). These SBs provide instructions to correct the antenna support installations and restore the structural capability of the affected areas.

For the reason described above, this AD requires modification of the RALT and GPS antenna installations.

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

**Modification(s):**

- (1) Before exceeding 2 600 flight cycles (FC) accumulated by the aeroplane since new (first flight), modify the aeroplane in accordance with the instructions of the SB1.
- (2) Before exceeding 3 100 flight cycles (FC) accumulated by the aeroplane since new (first flight), modify the aeroplane in accordance with the instructions of the SB2.

**Ref. Publications:**

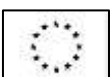
Airbus DS SB-295-34-0091C Revision 0 (original issue) dated 26 November 2025.

Airbus DS SB-295-34-0092C Revision 0 (original issue) dated 26 November 2025.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. This Proposed AD will be closed for consultation on 04 June 2026.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be



installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

4. For any question concerning the technical content of the requirements in this PAD, please contact: Airbus DS Services / Engineering Support, Fax: +34 91 585 3127,  
E-mail: [MTA.TechnicalService@airbus.com](mailto:MTA.TechnicalService@airbus.com).

For North American operators, contact alternatively:

E-mail: [TechnicalSupport@airbusmilitaryna.com](mailto:TechnicalSupport@airbusmilitaryna.com).

