



Notification of a Proposal to issue an Airworthiness Directive

PAD No.: 26-089

Issued: 24 July 2026

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

Design Change Approval Holder's Name: **Modification(s):**

ELBE FLUGZEUGWERKE GmbH

Airbus A330 Passenger to Freighter Conversion

Effective Date: [TBD - standard: 14 days after AD issue date]

STC Number(s): EASA Supplemental Type Certificate (STC) 10063798

Foreign AD: Not applicable

Supersedure: None

ATA 53 – Fuselage – Centre Fuselage – Modification

Manufacturer(s):

Airbus, formerly Airbus Industrie

Applicability:

Airbus A330 aeroplanes modified in accordance with EASA STC 10063798, serial numbers 600, 610, 709, 713, 717, 777, 781, 810, 815, 832, 854, 868, 879, 889, 937, 958, 964, 1107, 1124, 1401, 1453 and 1477.

Definitions:

For the purpose of this AD, the following definitions apply:

The SB: Elbe Flugzeugwerke GmbH (EFW) Service Bulletin EFW-SB-53-0086 Revision (Rev.) 04.

Reason:

It has been determined that certain brackets, which are installed on the C40 Web Spar to support the Passenger-to-Freighter (P2F) Bypass Ducts, have been installed with non-conforming fasteners. Fatigue analysis identified a risk that the affected fasteners may fail over time.



This condition, if not corrected, could affect the structural integrity of the affected area and, for aircraft equipped with centre tank capability, cause fuel leakage inside the centre fuselage.

To address this potential unsafe condition, EFW issued the SB, providing instructions to replace those fasteners.

For the reason described above, this AD requires modification of the affected aeroplanes.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Modification:

- (1) Within 4 months after the effective date of this AD, modify the aeroplane by replacing the fasteners in the area as identified in, and in accordance with the instructions of, the SB.

Credit:

- (2) Modification of an aeroplane accomplished before the effective date of this AD in accordance with the instructions of EFW SB EFW-SB-53-0086 original issue, Rev. 01, Rev 02 or Rev 03, is acceptable for compliance with the requirements of paragraph (1) of this AD for that aeroplane.

Ref. Publications:

EFW-SB-53-0086 original issue dated 28 June 2024, Rev. 01 dated 26 August 2024, Rev. 02 dated 05 September 2024, Rev. 03 dated 27 February 2026 and Rev. 04 dated 02 June 2026.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. This Proposed AD will be closed for consultation on 21 August 2026.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the [EU Aviation Safety Reporting System](#). This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA parts manufacturer approval (PMA), Supplemental Type Certificate (STC) or other modification.
4. For any question concerning the technical content of the requirements in this PAD, please contact EFW Office of Airworthiness, airworthiness@efw.aero.

