



# Notification of a Proposal to issue an Airworthiness Directive

**PAD No.: 26-111**

**Issued: 22 September 2026**

Note: This Proposed Airworthiness Directive (PAD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

In accordance with the EASA Continuing Airworthiness Procedures, the Executive Director is proposing the issuance of an EASA Airworthiness Directive (AD), applicable to the aeronautical product(s) identified below.

All interested persons may send their comments, referencing the PAD Number above, to the e-mail address specified in the 'Remarks' section, prior to the consultation date indicated.

**Design Approval Holder's Name:** **Type/Model designation(s):**

AIRBUS HELICOPTERS DEUTSCHLAND GmbH MBB-BK117 helicopters

**Effective Date:** [TBD - standard: 14 days after AD issue date]

**TCDS Number(s):** EASA.R.010

**Foreign AD:** Not applicable

**Supersedure:** None

## ATA 32 – Landing Gear – Emergency Flotation System Adhesive Label – Removal

### Manufacturer(s):

Airbus Helicopters (AH); Airbus Helicopters Deutschland GmbH; Kawasaki Heavy Industries Ltd.; and Airbus Helicopters Inc.

### Applicability:

MBB-BK117 D-2 (all variants) and D-3 helicopters, serial numbers 20001 to 20360 inclusive.

### Definitions:

For the purpose of this AD, the following definitions apply:

**The ASB:** AH Alert Service Bulletin (ASB) MBB-BK117-25-67-0006.

**Affected label:** Emergency Flotation System (EFS) adhesive label Part Number 214794-0.

**Groups:** Group 1 helicopters are those that have an affected label installed. Groups 2 helicopters are those that do not have an affected label installed.



**Reason:**

An investigation has revealed that a non-certified EFS adhesive label, displaying incorrect EFS gas cylinder pressure limits, was listed in the MBB-BK117 D-2 Illustrated Parts Catalogue and therefore may have been installed on the structure of the helicopter.

This condition, if not corrected, could lead to an incomplete inflation of the floats during ditching, potentially preventing timely evacuation from the helicopter and consequent injury to occupants.

To address this potential unsafe condition, AH issued the ASB, to provide instructions for removal of the affected label.

For the reason described above, this AD requires removal of those affected labels.

**Required Action(s) and Compliance Time(s):**

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

**Removal:**

- (1) For Group 1 helicopters: Within 18 months after the effective date of this AD, remove the affected label in accordance with the instructions of the ASB.

**Part(s) Installation:**

- (2) For Group 1 and Group 2 helicopters: From the effective date of this AD, do not install an affected label on any helicopter.

**Ref. Publications:**

AH ASB MBB-BK117-25-67-0006 original issue (Issue 001) dated 08 September 2026.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

**Remarks:**

1. This Proposed AD will be closed for consultation on 20 October 2026.
2. Enquiries regarding this PAD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu).
3. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this PAD, and which may occur, or have occurred on a product, part or appliance not affected by this PAD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this PAD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.



4. For any question concerning the technical content of the requirements in this PAD, please contact: Airbus Helicopters (Technical Support) at:  
Web portal: <https://airbusworld.helicopters.airbus.com> / Technical Requests Management, or  
E-mail: [TechnicalSupport.Helicopters@airbus.com](mailto:TechnicalSupport.Helicopters@airbus.com), or Telephone: +33 (0)4 42 859 789.

