**GSAC** 

# **AIRWORTHINESS DIRECTIVE**

## released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref.: 1996-105-043(B) R2 In case of any difficulty, reference should be made to the French original issue.

## **AIRBUS**

## A340 aircraft

Centre fuselage - Vertical beam Y=O at FR 53-2 (ATA 53)

## 1. APPLICABILITY:

AIRBUS A340 aircraft models -211, -212, -213, -311, -312 and -313, all serial numbers, except those which have received AIRBUS modification 42606 or AIRBUS Service Bulletin (SB) A340-53-4030.

## 2. REASONS

- **2.1.** In order to prevent a damage at an early stage at frame 53-2 on the vertical support beam which supports the CLG actuator attachment fitting, caused by cracks evidenced during fatigue tests and which could lead to a reduced structural integrity.
- **2.2.** The Revision 1 of this Airworthiness Directive (AD) introduced in paragraph 3.2.1., as an Alternative Mean Of Compliance (AMOC), the accomplishment of the instructions given in SB AIRBUS A340-53-4030.
- **2.3.** The Revision 2 of this AD consists of revising the "Applicability" paragraph which limits AIRBUS A340 aircraft models as listed in the paragraph 1.

#### 3. COMPLIANCE:

The following measures are rendered mandatory from the effective date of this AD at original issue.

- **3.1.** Before accumulation of 6,400 flight cycles, perform a rototest inspection of the beam at the upper first fastener row of the assembly with CLG retraction fitting in accordance with the instructions given in AIRBUS SB A340-53-4043.
- **3.2.** Depending upon the results of this inspection:
  - 3.2.1. either if no crack is found:

repeat the inspection in accordance with the instructions given in AIRBUS Service Bulletin A340-53-4043 at intervals not exceeding 4,700 flight cycles,

or

accomplish the preventive measures in accordance with the instructions given in AIRBUS SB A340-53-4030.

.../...

n/CG

**GSAC** 

AIRWORTHINESS DIRECTIVE

1996-105-043(B) R2

Page n°

2

**3.2.2.** or if a crack is found and depending on the dimension measured: refer to the action corresponding to the thresholds defined on the synoptic chart 1 of AIRBUS SB A340-53-4043.

ref.:

**NOTE:** No further action is required after accomplishment in accordance with AIRBUS SB A340-53-4030.

REF.: AIRBUS Service Bulletin A340-53-4043 AIRBUS Service Bulletin A340-53-4030

(Any later approved revision of these SBs is acceptable).

\_\_\_\_

This Revision 2 replaces AD 96-105-043(B) R1 dated July 30, 1997.

\_\_\_\_\_

# **EFFECTIVE DATES**:

Original AD : JUNE 15, 1996
Revision 1 : AUGUST 09, 1997
Revision 2 : MARCH 01, 2003