

# AIRWORTHINESS DIRECTIVE

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*Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.*

Translation of 'Consigne de Navigabilité' ref. : 1997-363-076(B) R3  
In case of any difficulty, reference should be made to the French original issue.

## AIRBUS INDUSTRIE

### A340 aircraft

Centerline gear - Inspection of the CLG RH inboard lug (ATA 32)

#### APPLICABILITY :

This Airworthiness Directive (AD) applies to AIRBUS INDUSTRIE A340 aircraft models -211 ; -212 ; -213 ; -311 ; -312 and -313, all serial numbers which have not received AIRBUS INDUSTRIE modification (mod.) 45302 (AIRBUS INDUSTRIE Service Bulletin (SB) A340-32-4097) nor mod. 46880 (SB A340-32-4138).

#### REASONS :

CLG RH main fitting inboard attachment lugs found cracked or ruptured have been reported on in-service aircraft.

The rupture of the CLG RH main fitting inboard attachment lug may lead in the worst case to the CLG becoming detached during taxiing, at take-off or at landing and potentially cause a risk to people and property on the ground.

Revision 3 of this AD introduces the mod. 46880 (installation of CLG std "F") as an alternate to the modification 45302 in the paragraph "APPLICABILITY".

#### COMPLIANCE :

The following measures are rendered mandatory:

##### Configuration 1 :

For in-service aircraft which have not received either mod. 43028 or SB A340-32-4083 and which centerline gears have accumulated 150 flight cycles or more :

- 1.1. Within 7 days at the latest from effective date of the original issue of this AD, perform a detailed visual inspection for cracks of the main fitting in accordance with instructions given in SB A340-32-4091.
- 1.2. Depending upon the inspection results, carry out the specified instructions in accordance with the flow chart (figure 1) as indicated in the SB A340-32-4091.

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December 29, 1999

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**Configuration 2 :**

For in-service aircraft which have received either mod. 43028 or SB A340-32-4083 and which centerline gears have accumulated 1020 flight cycles and more :

- 2.1. Within 7 days at the latest from effective date of the original issue of this AD, perform a detailed visual inspection for cracks of the main fitting in accordance with instructions given in SB A340-32-4091.
- 2.2. Depending upon the inspection results, carry out the specified instructions in accordance with the flow chart (figure 1) as indicated in SB A340-32-4091.

**Note** : Inspection SB A340-32-4091 is still applicable after accomplishment of SB A340-32-4105.

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**REF.** : AIRBUS INDUSTRIE Service Bulletin A340-32-4091  
(or any further approved revision)  
AIRBUS INDUSTRIE Service Bulletin A340-32-4097  
(or any further approved revision)

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This Revision 3 replaces the Airworthiness Directive 97-363-076(B) R2 dated July 15, 1998.

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**EFFECTIVE DATES:**

**Original AD : NOVEMBER 29, 1997**  
**Revision 1 : MARCH 21, 1998**  
**Revision 2 : JULY 25, 1998**  
**Revision 3 : JANUARY 08, 2000**