

AIRWORTHINESS DIRECTIVE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

98-022-062(B) R1

In case of any difficulty, reference should be made to the French original issue.

AIRBUS INDUSTRIE

A330 Aircraft

ATA 36 - Wing bleed leak detection

APPLICABILITY :

AIRBUS INDUSTRIE A330 aircraft, models -301, -321, -322, -341 and -342 which have not received either AIRBUS INDUSTRIE modification 46115 or AIRBUS INDUSTRIE Service Bulletin A330-36-3012.

REASONS :

A wiring cross connection has been discovered on an in-service aircraft which inhibited wing bleed leak ECAM warning detection.

In case of a bleed air leak in the wing area, the automatic closure of the associated engine bleed air system would not be performed and the flight would not be informed by an ECAM warning display.

The consequences would be an undetected long exposure to hot air of the environment close to the leak location.

COMPLIANCE :

Rendered mandatory from effective date of this Airworthiness Directive, unless accomplished previously :

- Not later than 500 flight hours from effective date of this Airworthiness Directive accomplish the instructions as given in AIRBUS INDUSTRIE AOT (All Operator Telex) 36-04 dated December 08, 1997 or AIRBUS INDUSTRIE Service Bulletin A330-36-3012.

Note : The instructions specified in AOT 36-04 dated December 08, 1997 and those of AIRBUS INDUSTRIE Service Bulletin A330-36-3012 are equivalent.

June 17, 1998

**AIRBUS INDUSTRIE
A330 Aircraft**

98-022-062(B) R1

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Ref. : AIRBUS INDUSTRIE AOT 36-04 dated December 08, 1997
AIRBUS INDUSTRIE Service Bulletin A330-36-3012
(or any further approved revision)

This Revision 1 replaces the original AD 98-022-062(B) dated January 28, 1998.

EFFECTIVE DATES :

Original AD : FEBRUARY 07, 1998
Revision 1 : JUNE 27, 1998