**GSAC** 

# AIRWORTHINESS DIRECTIVE

### released by DIRECTION GENERALE DE L'AVIATION CIVILE

Inspection and/or modifications described below are mandatory. No person may operate a product to which this Airworthiness Directive applies except in accordance with the requirements of this Airworthiness Directive.

Translation of 'Consigne de Navigabilité' ref.: 98-277-040(A) R1 In case of any difficulty, reference should be made to the French original issue.

#### **EUROCOPTER**

## SA 315 Helicopters

Main Rotor Blades - Bonding of Blade Skin and Blade Root Reinforcement Strip

## 1. EFFECTIVITY

This Airworthiness Directive applies to SA 315B helicopters fitted with main rotor blades bearing the following part numbers:

- 3160S11-10000 all part numbers, - 3160S11-30000 - all part numbers, - 3160S11-35000 - all part numbers, - 3160S11-40000 - all part numbers. - 3160S11-45000 - all part numbers, - 3160S11-50000 - all part numbers,
- 3160S11-55000 all part numbers.

#### 2. REASON

This Airworthiness Directive is issued following the failure of a main rotor blade on a SE 3130 helicopter.

# 3. COMPLIANCE

The actions defined in paragraph 4 must be undertaken:

- Within 25 flying hours or 6 months (at the first limit reached), from the effective date of the original issue of this Airworthiness Directive (unless already accomplished).
- 3.2. Then every 100 flying hours or every 6 months (at the first limit reached).

In case of separation inside the non-hatched area of the strip, the check interval is reduced to 25 flying hours.

**3.3.** Before putting into service a blade held as spares.

#### 4. MANDATORY ACTIONS

Check the spar skin and the blade root reinforcement strip area for separations, cracks and corrosion, in accordance with the procedure defined in paragraph 2. of referenced SA 315 Service Bulletin No. 05.34.

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**EUROCOPTER December 16, 1998 SA 315 Helicopters** 

98-277-040(A) R1

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