	AIRWORTHINESS DIRECTIVE No F-2002-518 R1		Distribution: A	Issue date: February 18, 2004 <i>Correction of November 21, 2007</i>	Page : 1/2
	This Airworthiness Directive is published by the DGAC : <div style="display: flex; align-items: flex-start;"> <div style="margin-right: 10px;"> <input checked="checked" type="checkbox"/> </div> <div> on behalf of EASA, the Primary Airworthiness Authority for the affected product. </div> </div> <div style="display: flex; align-items: flex-start; margin-top: 5px;"> <div style="margin-right: 10px;"> <input type="checkbox"/> </div> <div> as the Registration Airworthiness Authority for the affected aircraft.. </div> </div>			<i>Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.</i>	
GSAC publication	No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.				
Corresponding foreign Airworthiness Directive(s): Not applicable			Airworthiness Directive(s) replaced: 2002-518 original issue		
Person in charge of airworthiness: ISSOIRE AVIATION			Type(s): Wassmer WA 50 aeroplanes		
Type certificate(s) No. 51 TCDS No 126					
ATA chapter: 55		Subject: Stabilizers - Horizontal stabilizer			

Correction : the corrected lines are marked with the letter "C" in the margin.

1. EFFECTIVITY:

C WASSMER WA 50 aeroplanes models WA 51, WA 51A, WA 52 and WA 54, serial numbers from 11 and
C up, and serial numbers 1 through 10 on which the horizontal stabilizer (P/N WA 51-30-10) has been replaced.


2. REASONS:

The purpose of this Airworthiness Directive (AD) is to preclude the appearance of delamination between the skins and the core of the sandwich structure of the horizontal stabilizer.

This revision 1 specifies the application conditions of § 3 "Mandatory actions and compliance time".

3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

- 3.1.** If not already accomplished, during the next scheduled maintenance visit or within 50 flight hours following the effective date of this AD (whichever occurs first), inspect the horizontal stabilizer according to the instructions of the referenced Service Bulletin (SB).
- 3.2.** If no delamination is found or if the delaminations are within the acceptable margins defined in the referenced SB, repeat the inspection of § 3.1. at intervals not exceeding 110 flight hours or 12 months (whichever occurs first).
- 3.3.** If delaminations are out of the acceptable margins, replace the affected skin panels before the next flight according to ISSOIRE AVIATION repair instructions.
- 3.4.** The repetitive inspection of § 3.2. is not longer required for skin panels which have been replaced. The complete replacement of all skin panels cancels the repetitive inspection of § 3.2. here above.

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4. REFERENCE PUBLICATION:

ISSOIRE AVIATION Service Bulletin No 40a dated December 16, 2003.

5. EFFECTIVE DATES:

Original issue : October 26, 2002
Revision 1 : February 28, 2004.

6. REMARK:

For questions concerning the technical contents of this AD requirements, contact:
ISSOIRE AVIATION – Customer support – Facsimile: 33 (0)4 73 89 54 59.

7. APPROVAL:

This AD is approved under EASA reference No 2004-1051 dated February 10, 2004.