	AIRWORTHINESS DIRECTIVE No F-2004-124	Distribution: B	Issue date: July 21, 2004	Page : 1/2
Direction générale de l'aviation civile France	This Airworthiness Directive is published by the DGAC: <input type="checkbox"/> on behalf of EASA, the Primary Airworthiness Authority for the affected product. <input checked="" type="checkbox"/> as the Registration Airworthiness Authority for the affected aircraft.		<i>Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.</i>	
GSAC publication	<p style="text-align: center;">No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.</p>			
Corresponding foreign Airworthiness Directive(s): Not applicable		Airworthiness Directive(s) replaced: 2000-317-045 cancelled by its Revision 1		
Person in charge of airworthiness: AIRBUS SAS		Type(s): A300-600ST BELUGA aircraft		
Type certificate(s) No. None TCDS No None				
ATA chapter: 57	Subject: Wing - Centre wing box - FR47 angle fittings			

1. EFFECTIVITY:

AIRBUS A300-600ST BELUGA aircraft, all certified serial numbers.

2. REASONS:

This Airworthiness Directive (AD) is issued subsequent to the publication of a new repetitive inspection program for fuselage frame 47 at certain fasteners of the wing box angle fitting.

This new inspection program specific to the BELUGA aircraft replaces the one mandated by AD 2000-317-045 and follows on from the results of the inspections performed on A300 and A300-600 aircraft and is modified to especially:

- extend this inspection program to certain holes in the flange (horizontal face) of the angle fitting,
- extend the inspection to other holes in the web (vertical face) of the angle fitting,
- integrate a redefinition of the inspection intervals of all holes in the web of the angle fitting,
- include both Flight Cycles/Flight Hours threshold and interval data.


3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

The following measures are rendered mandatory from the effective date of this AD.

3.1. Inspection program for the web (vertical face) of the angle fitting:

3.1.1. At the threshold and in accordance with the instructions defined in AIRBUS Service Bulletin (SB) A300-57-9001, perform an inspection of holes H, I, K, L, M, N, Y, U, V, W and X.

3.1.2. According to the results of the previous inspection and according to the corrective measures undertaken, repeat the inspection in accordance with the instructions and at the intervals defined in SB A300-57-9001.

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3.2. Inspection program for the flange (horizontal face) of the angle fitting:

3.2.1. At the threshold and in accordance with the instructions defined in AIRBUS Service Bulletin (SB) A300-57-9002, perform an inspection of holes A, B, C, D, E, F, G, P, Q, S and T.

3.2.2. According to the results of the previous inspection and according to the corrective measures undertaken, repeat the inspection in accordance with the instructions and at the intervals defined in SB A300-57-9002.

3.3. The aircraft manufacturer must be informed of the inspection results whatever they are.

4. REFERENCE PUBLICATIONS:

Bulletin Service AIRBUS A300-57-9001
Bulletin Service AIRBUS A300-57-9002
(Any further approved revision of these SBs is acceptable).

5. EFFECTIVE DATE:

July 31, 2004.

6. REMARK:

For questions concerning the technical contents of this AD's requirements, contact:

AIRBUS SAS - Didier AURICHE - Fax : 33 5 61 93 45 80.

7. APPROVAL:

This AD is approved by DGAC.