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|  | AIRWORTHINESS DIRECTIVE No F-2004-148 | Distribution: A | Issue date: September 01, 2004 | Page : 1/2 |
| | Direction générale de l'aviation civile France GSAC publication | This Airworthiness Directive is published by the DGAC on behalf of EASA, Airworthiness Authority of the State of Design for the affected product, part or appliance. | | <i>Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.</i> |
| No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry. | | | | |
| Corresponding foreign Airworthiness Directive(s): Not applicable | | Airworthiness Directive(s) replaced: None | | |
| Person in charge of airworthiness: EADS SOCATA | | Type(s): TB 21 aeroplanes | | |
| Type certificate(s) No. 165 TCDS No 165 | | | | |
| ATA chapter: 56 | Subject: Windows-Windshield inspection | | | |

1. EFFECTIVITY:

EADS SOCATA TB 21 aeroplanes, from serial number (SN) 500 to SN 1925 inclusive.

2. REASON:

A crack initiation at an attachment fitting may lead to a in-flight windshield cracking.

3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

The following measures are made mandatory from the effective date of this Airworthiness Directive (AD):

- a) Within the next 110 flight hours or in the next 12 months, whichever occurs first;
- b) Then in a recurring way, not to exceed 13 months;


Perform the windshield inspection, repair or replacement as necessary, in accordance with the Service Bulletin (SB) EADS SOCATA No 10-144-56 instructions.

4. REFERENCE PUBLICATION:

SB EADS SOCATA No 10-144-56 dated June 2004.

5. EFFECTIVE DATE:

September 11, 2004.

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6. REMARKS:

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7. APPROVAL:

This AD is approved under EASA reference No 2004-9023 dated August 24, 2004.