	<b>AIRWORTHINESS DIRECTIVE</b> <b>No F-2004-034</b>		Distribution: <b>A</b>	Issue date: <b>March 17, 2004</b>	Page : <b>1/2</b>
	Direction générale de l'aviation civile France  GSAC publication	This Airworthiness Directive is published by the DGAC : <input checked="" type="checkbox"/> on behalf of EASA, the Primary Airworthiness Authority for the affected product. <input type="checkbox"/> as the Registration Airworthiness Authority for the affected aircraft..		<i>Translation of « Consigne de Navigabilité » of same number.          In case of difficulty, reference should be made to the French issue.</i>	
<b>No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.</b>					
Corresponding foreign Airworthiness Directive(s): <b>Not applicable</b>			Airworthiness Directive(s) replaced: <b>None</b>		
Person in charge of airworthiness: <b>EUROCOPTER</b>			Type(s): <b>EC 130 helicopters</b>		
Type certificate(s) No. <b>84</b> TCDS No <b>157</b>					
ATA chapter: <b>28</b>		Subject: <b>Fuel - Fuel bleed lever</b>			

**1. EFFECTIVITY:**

EC 130 B4 helicopters pre-MOD 073239.

**2. REASON:**

This Airworthiness Directive (AD) is issued following some cases of loss of the fuel bleed lever in flight, on AS 350 helicopters. which could have led to a collision with the tail rotor.

**3. MANDATORY ACTIONS AND COMPLIANCE TIMES:**

The following measures are rendered mandatory from the effective date of this AD:


- 3.1. No later than within 6 months, remove and modify the fuel bleed lever in compliance with the instructions described in paragraphs 2.B and 2.C of referenced EUROCOPTER EC 130 Alert Service Bulletin (ASB) No. 28A001.
- 3.2. Before installation on an aircraft of a fuel bleed lever held as spares, comply with the instructions described in paragraphs 2.B and 2.C of the referenced ASB.

**4. REFERENCE PUBLICATION:**

EUROCOPTER EC 130 Alert Service Bulletin No. 28A001.  
 (Any subsequent approved revision of this ASB is acceptable).

**5. EFFECTIVE DATE:**

Upon receipt from March 17, 2004.

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**6. REMARK:**

For any questions concerning the technical content of the requirements in this AD, please contact:

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**7. APPROVAL:**

This AD is approved under EASA reference No 2004-2106 dated March 09, 2004.