	All	RWORTHINESS DIRECTIVE	Distribution:	Issue date:	Page :
	No F-2004-073		В	May 26, 2004	1/2
Direction générale de l'aviation civile France	This Airworthiness Directive is published by the DGAC: On behalf of EASA, the Primary Airworthiness Authority for the affected product. as the Registration Airworthiness Authority for the affected aircraft			Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.	
GSAC publication	No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.				
Corresponding foreign Airworthiness Directive(s):			Airworthiness Directive(s) replaced:		
Not applicable			None		
Person in charge of airworthiness: AIRBUS SAS			Type(s): A318, 319, A320 and A321 aircraft		
Type certificate(s) No. 180					
TCDS No 180					
ATA chapter:		Subject:			
28	Fuel - Prevention against explosion risks - Adjacent systems				

1. **EFFECTIVITY**:

AIRBUS A318, A319, A320 and A321 aircraft, all certified models, all serial numbers, except aircraft that have received AIRBUS modification 33518 in production or AIRBUS Service Bulletin (SB) A320-92-1032 in service.

2. REASONS:

Further to the accident of a Boeing 747-131 aircraft (flight TWA800), the FAA has published SFAR 88 (Special Federal Aviation Regulation 88).

In their letters referenced 04/00/02/07/01-L296, dated March 04, 2002 and 04/00/02/07/03-L024, dated February 03, 2003, the JAA recommended the application of a similar regulation to the National Aviation Authorities (NAA).

Under this regulation, all holders of type certificates for passenger transport aircraft, with either a passenger capacity of 30 or more or a payload capacity of 7,500 pounds (3,402 kg) or more, which have received their certification since January 01, 1958, are required to conduct a design review against explosion risks.

This Airworthiness Directive (AD), which renders mandatory the removal of the maintenance light with Part Number (PN) 2LA002606-01 and its electrical connections in the hydraulics bay, is a consequence of this design review.

3. MANDATORY ACTION AND COMPLIANCE TIME:

No later than December 31, 2005, remove the maintenance light with PN 2LA002606-01 in the hydraulics bay and disconnect the wiring in accordance with the instructions of SB A320-92-1032.



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4. REFERENCE PUBLICATION:

Service Bulletin AIRBUS A320-92-1032 (Any later approved revision of this document is acceptable).

5. EFFECTIVE DATE:

June 05, 2004.

6. REMARK:

For questions concerning the technical contents of this AD's requirements, contact:

AIRBUS SAS - Bureau de Navigabilité - EAS - Fax 33 5 61 93 44 51.

7. APPROVAL:

This AD is approved under EASA reference No 2004-5336 dated May 17, 2004.