	AIRWORTHINESS DIRECTIVE No F-2005-011		Distribution: B	Issue date: January 19, 2005	Page : 1/2
	Direction générale de l'aviation civile France GSAC publication	This Airworthiness Directive is published by the DGAC on behalf of EASA, Airworthiness Authority of the State of Design for the affected product, part or appliance.		<i>Translation of « Consigne de Navigabilité » of same number. In case of difficulty, reference should be made to the French issue.</i>	
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.					
Corresponding foreign Airworthiness Directive(s): Not applicable			Airworthiness Directive(s) replaced: None		
Person in charge of airworthiness: AIRBUS SAS			Type(s): A340-200/-300 aircraft		
Type certificate(s) No. A.015 TCDS No A.015					
ATA chapter: 54	Subject: Nacelles/Pylons - Engine CFM56-5C pylon - Inspection of pyramid forward fitting spotfacing				

1. EFFECTIVITY:

AIRBUS A340 aircraft, models -211, -212, -213, -311, -312 and -313, all serial numbers up to 0352 inclusive.

2. REASONS:

During a routine inspection, an operator observed on two aircraft of his fleet, a machining non conformity of the spotfacing on two pylon/engine attachment located on the pyramid forward fitting.

The diameter of the spotfacing was less than the nominal diameter, thus the head of the pylon/engine attachment bolt did not bear correctly on the surface of the spotfacing.


This situation, if not corrected, could reduce the structural integrity of the pylon/engine attachment bolt.

3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

The following actions are rendered mandatory from the effective date of this AD unless already accomplished:

- 3.1. Within 18 months at latest from the effective date of this AD, inspect, on the pyramid forward fitting of the four engine pylons, the spotfacing of the attachment located at the front of the fitting in accordance with the instructions of AIRBUS Service Bulletin (SB) A340-54-4009.
- 3.2. If the results of the inspection show that the spotfacing is satisfactory, no further action is required in accordance with the instructions of SB A340-54-4009.
- 3.3. If the diameter of the spotfacing is not in compliance with the nominal diameter, before next flight, machine the spotfacing and replace the corresponding pylon/engine attachment components by new ones in accordance with the instructions of SB A340-54-4009.

Note: No further action is required after inspection and machining, if necessary, of the pyramids of the four pylons.

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4. REFERENCE PUBLICATION:

AIRBUS Service Bulletin A340-54-4009
Any further approved revision of this SB is acceptable.

5. EFFECTIVE DATE:

January 29, 2005.

6. REMARK:

For questions concerning the technical contents of this AD's requirements, contact:
AIRBUS SAS – Airworthiness Office - EAL - Fax : 33 5 61 93 45 80.

7. APPROVAL:

This AD is approved under EASA reference No 2005-495 dated January 12, 2005.