K	AIRWORTHINESS DIR No F-2005-134		Distribution:	Issue date: August 03, 2005	Page : 1/2	
Direction générale de l'aviation	This Airworthiness Directive is published by the DO EASA, Airworthiness Authority of the State of Desi product, part or appliance.					
civile France GSAC publication	No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive, unless otherwise agreed with the Authority of the State of Registry.					
Correspondir	ng foreign Airworthiness Directive(s):	Airwort	Airworthiness Directive(s) replaced:			
Not applicable		None	None			
Person in charge of airworthiness: EADS SOCATA			Type(s): TBM 700 aircraft			
Type certifica	te(s) No. EASA.A.010					
TCDS No E	ASA.A.010					
ATA chapter:	Subject:	ct:				
53	Fuselage - Pilot door locking fittings					

1. EFFECTIVITY:

This Airworthiness Directive (AD) is applicable to EADS SOCATA TBM 700 B and C variants, from serial number 126 to 322 inclusive and equipped with a pilot door.

2. REASON:

A non-respect of the pilot door adjustment procedure may damage the stop fittings and could result in a consequent depressurization of the airplane.

3. MANDATORY ACTIONS AND COMPLIANCE TIMES:

The following measures are rendered mandatory from the effective date of this AD, unless previously accomplished:

- **3.1.** Within the next 10 flight hours accomplish a detailed inspection of the locking stops in accordance with the accomplishment instructions of the Service Bulletin EADS SOCATA given in reference.
- **3.2.** If no faulty fitting is found, no further action is required.
- **3.3.** If any faulty fitting is found, replace damaged fitting with new one. Pressurized flights will be forbidden until damaged fittings are replaced.

These actions must be done in accordance with the manufacturer's technical Publications.

4. REFERENCE PUBLICATION:

Service Bulletin EADS SOCATA TBM 700 n° 70-131 dated July 2005 (Any subsequent approved revision of this document is acceptable).



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5. **EFFECTIVE DATE**:

On receipt as from August 03, 2005.

6. REMARK:

For any questions concerning the technical content of the requirements in this AD, please contact:

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7. APPROVAL:

This AD is approved under EASA reference No 2005-6121 dated July 27, 2005.

