EASA	AIRW	ORTHINESS DIRECTIVE
X	AD No.: 2006 – 00)94
	Date: 21 April 200	06
		iness Directive applies, except in accordance with the erwise agreed with the Authority of the State of Registry.
Type Approval He	older's Name:	Type/Model designation(s):
EUROCOPTER		AS 350
TCDS Number: EAS	SA. R. 008	
Foreign AD: None.		
Supersedes: None.		
ATA 76	Engine Controls – Tw	ist Grip Assembly
Manufacturer(s):	EUROCOPTER	
Applicability:		s, equipped with twist grips before embodiment of vist grip assembly compatible with the ARRIEL 2B1
Reason:	This Airworthiness Directive (AD) is issued following a case experienced during an autorotation training procedure during which the engine remained at idle rating although the twist grip had been turned back to the "FLIGHT" position.	
		I jamming of the "forced idle" microswitch (called t below) pin in the pushed-in position.
	The jamming results i	n the engine speed being held at "forced idle":
	 when the pilot turns completion of autorota 	the twist grip back to the "FLIGHT" position on ation training,
	or,	
	 when the pilot turns for governor failure. 	the grip in the low flow rate direction during training
Effective Date:	05 May 2006	

following measures are rendered mandatory as from the effective of this AD: o later than within 110 flying hours, adjust the microswitch and neck it for correct operation, in compliance with the instructions becified in paragraphs 2.B.2. and 2.B.3. of referenced JROCOPTER AS 350 Alert Service Bulletin (ASB) No. 05.00.49. eplace the microswitch in compliance with the instructions specified paragraph 2.B.4. of the referenced ASB: For helicopters with less than 440 flying hours: at the latest at 550 lying hours, carrying out the checks defined in paragraph 3. of this AD before the helicopter has logged 550 flying hours. For helicopters with 440 flying hours or more: no later than within 10 flying hours. arry out a functional test of the micro switch every 110 flying hours, compliance with the instructions specified in paragraph 2.B.3. of the ferenced ASB.
 beck it for correct operation, in compliance with the instructions becified in paragraphs 2.B.2. and 2.B.3. of referenced JROCOPTER AS 350 Alert Service Bulletin (ASB) No. 05.00.49. beplace the microswitch in compliance with the instructions specified paragraph 2.B.4. of the referenced ASB: For helicopters with less than 440 flying hours: at the latest at 550 lying hours, carrying out the checks defined in paragraph 3. of this AD before the helicopter has logged 550 flying hours. For helicopters with 440 flying hours or more: no later than within 10 flying hours. arry out a functional test of the micro switch every 110 flying hours, compliance with the instructions specified in paragraph 2.B.3. of the ferenced ASB.
paragraph 2.B.4. of the referenced ASB: For helicopters with less than 440 flying hours: at the latest at 550 lying hours, carrying out the checks defined in paragraph 3. of this AD before the helicopter has logged 550 flying hours. For helicopters with 440 flying hours or more: no later than within 10 flying hours. arry out a functional test of the micro switch every 110 flying hours, compliance with the instructions specified in paragraph 2.B.3. of the ferenced ASB.
lying hours, carrying out the checks defined in paragraph 3. of this AD before the helicopter has logged 550 flying hours. For helicopters with 440 flying hours or more: no later than within 10 flying hours. arry out a functional test of the micro switch every 110 flying hours, compliance with the instructions specified in paragraph 2.8.3. of the ferenced ASB.
110 flying hours. arry out a functional test of the micro switch every 110 flying hours, compliance with the instructions specified in paragraph 2.B.3. of the ferenced ASB.
compliance with the instructions specified in paragraph 2.B.3. of the ferenced ASB.
eplace the microswitch every 550 flying hours in compliance with the structions specified in paragraph 2.B.4 of the referenced ASB.
OCOPTER AS 350 Alert Service Bulletin No. 05.00.49,
ter approved revisions.
requested and appropriately substantiated the responsible EASA anager for the related product has the authority to accept Alternativ ethod of Compliance (AMOCs) for this AD.
his AD was posted as PAD 06-043 for consultation on 23 February 006 with a comment period until 09 March 2006. No comment was ised during the consultation period.
nquiries regarding this Airworthiness Directive should be referred to r. M. Capaccio, Airworthiness Directive Focal Point - Certification rectorate, EASA. E-mail: <u>ADs@easa.eu.int</u>
or any questions concerning the technical content of the quirements in this AD, please contact: EUROCOPTER (STXI) éroport de Marseille Provence 13725 Marignane Cedex - France n. : 33 (0) 4 42 85 97 97 - Fax : 33 (0) 4 42 85 99 66. mail : <u>Directive.technical-support@eurocopter.com</u>