


EASA	AIRWORTHINESS DIRECTIVE
	AD No.: 2006 – 0129 Date: 22 May 2006
No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Registry.	
Type Approval Holder's Name: AIRBUS	Type/Model designation(s): A330 aircraft
TCDS Number: EASA.A.004	
Foreign AD: N/A	
Supersedure: F-2005-069(B), EASA approval No. 2005-3631	
ATA 05	Time Limits - Safe Life Airworthiness Limitation Items – ALS Part 1
Manufacturer(s):	AIRBUS (formerly AIRBUS INDUSTRIE)
Applicability:	AIRBUS A330 aircraft, all certified models, all serial numbers.
Reason:	<p>The list of Safe Life Airworthiness Limitation Items (SL ALI) was given in the AIRBUS A330 Maintenance Planning Document (MPD) Section 9-1 (Life Limits/Monitored parts) up to revision 05. The content of the MPD Section 9 has been transferred into the ALS (Airworthiness Limitations Section) and the limitations applicable to SL ALI are now listed in the ALS Part 1, which is approved by the European Aviation Safety Agency (EASA).</p> <p>The revision 00 of the AIRBUS A330 ALS Part 1 was issued to introduce the Enhanced main landing gear, the A330-203 weight variants 054 and 055, the A330-243 weight variants 026 and 027, the A330-342 weight variant 052, the Gear support rib, the Forward pintle fitting, the Retraction jack fitting and the Side stay fitting. Further, the values of some limitations already listed were decreased.</p>
Effective Date:	06 June 2006

Compliance:	<ol style="list-style-type: none"> 1. Except as provided in paragraph 3 of this AD, it is mandatory to strictly adhere to the life values of the parts as defined in the sub-parts 1-2 and 1-3 of the A330 ALS Part 1 revision 00 from 06 August 2006. 2. From the effective date of this AD to 05 August 2006, it is mandatory to strictly adhere to the life values of the parts as defined, either: <ul style="list-style-type: none"> – In the sub-sections 9-1-2 and 9-1-3 of the A330 MPD Section 9-1 at revision 05, except as provided in paragraph 3 of this AD, or, – In the sub-parts 1-2 and 1-3 of the A330 ALS Part 1 at revision 00. 3. From the effective date of this AD, it is mandatory to strictly adhere to the life values defined in the sub-part 1-2 of the A330 ALS Part 1 revision 00 for: <ul style="list-style-type: none"> – The main landing gear retraction bracket PN 201428332. <p>Note: It is necessary to implement and maintain traceability as defined in the A330 ALS Part 1.</p>
Ref. Publications:	<p>A330 MPD Section 9-1, revision 05 approved on 07-Apr-2005. A330 ALS Part 1, revision 00 approved on 23 March 2006. or later approved revisions of the A330 ALS Part 1.</p>
Remarks:	<ol style="list-style-type: none"> 1. If requested and appropriately substantiated the responsible EASA manager for the related product has the authority to accept Alternative Method of Compliance (AMOCs) for this AD. 2. This AD was posted as PAD 06-075 for consultation on 29 March 2006 with a comment period until 12 April 2006. No comment was raised during the consultation period. 3. Enquiries regarding this AD should be addressed to Mr. M. Capaccio, AD Focal Point, Certification Directorate, EASA. E-mail ADs@easa.europa.eu . 4. For any question concerning the technical content of the requirements in this AD, please contact AIRBUS SAS - EAW (Airworthiness Office, Ph.: + 33 5 61 93 36 96; Fax: + 33 5 61 93 44 51)