EASA AIRWORTHINESS DIRECTIVE AD No.: 2006 - 0165 Date: 13 June 2006 No person may operate an aircraft to which an Airworthiness Directive applies, except in according ce with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the State of Regist Type/Model design ation//: Type Approval Holder's Name: **AIRBUS** A318, A319 and . aircraft 102 TCDS Number: EASA.A.064 Foreign AD: None Supersedure: F-2004-018 (EASA approval No 2004-752) **ATA 05** Airworthings Limitation nems (ALI) – Structural maintenance US, (famerly ARBUS INDUSTRIE) Manufacturer(s): 9, A320 and A321 aircraft all certified models, all serial Applicability: mbers, except A319 aircraft that have received AIRBUS modifications 28238, 28162 and 28342 in production AIRBUS A318-111 and A318-112 aircraft, all serial numbers. The certification requirements for damage-tolerant structure were formerly (eason: provided in AIRBUS A318/A319/A320/A321 Maintenance Planning Document (MPD) subsection 9-2 (Airworthiness Limitations Items - ALI). The content of the MPD section 9 has been transferred into the ALS (Airworthiness Limitations Section) and the ALI are now listed in the ALS Part 2 that is approved by the European Aviation Safety Agency (EASA). ALS Part 2 refers to A318/A319/A320/A321 ALI document AI/SE-M4/95A.0252/96 that has been approved at issue 7 on February 7th, 2006, to reduce threshold and/or interval of some existing ALI and introduce some new ALI.

This Airworthiness Directive (AD) introduces a limited exemption of 2 weeks to comply with the requirements of A318/A319/A320/A321 ALI document

AI/SE-M4/95A.0252/96 Issue 7.

Effective Date:	16 June 2006
Compliance:	 From the effective date of this AD to June 30, 2006, it is mandatory to adhere to the requirements of the A318/A319/A320/A321 ALI document AI/SE-M4/95A.0252/96 at either:
	issue 6 approved on July 15, 2003 or
	 issue 7 approved on February 7, 2006
	2. From June 30, 2006, it is mandatory to adhere to the require defined in the A318/A319/A320/A321 ALI document AI/SF M4/95A.0252/96 at issue 7 approved on February 7, 200
Ref. Publications:	AIRBUS AI/SE-M4/95A.0252/96 issue 6 approved op July 15, 200, by EASA.
	AIRBUS AI/SE-M4/95A.0252/96 issue 7 approved on a gruary 7 2006 by EASA.
	or later approved revisions.
Remarks :	If requested and appropriately substantiates the responsible EASA manager for the related project has the authority to accept Alternative Method of Compliance (AMO s) for this AP.
	2. This AD was posted to PAD 06-0-2R11 or consultation on 11 May 2006 with a comment period until to May 2006. The Comment Response Document can be found in nttp://nad.easa.eu.int/ .
	3. Enquiries recaron, this AD should be addressed to Mr. M. Capaccio, AD Focal Poir r, Certification Directorate, EASA. E-mail A. @er
	4. For any question concerning the technical content of the requirements in this AD. Tease intact AIRBUS - Fax 33 5 61 93 44 51