EASA

REVISED EMERGENCY AIRWORTHINESS DIRECTIVE



EAD No.: 2006 - 0258 R1-E

Date: 29 August 2006

No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the state of registry

Type Approval Holder's Name:

Type/Model designation(s):

EUROCOPTER

AS 365 N 3A 366 and EU 55 series

TCDS Number: DGAC 159

Foreign AD: Not applicable

Supersedes: EASA EAD 2006-0258-E (original issue with fawn) and 2006-0051-E

ATA 64

Tail Rotor - Tail Searbox Oil Level and Tail Rotor Pitch Control Rod Bearing - Inspection/Replacement

Manufacturer(s):

OPTEM Formerly EUROCOPTER FRANCE – AEROSPATIALE)

Applicability:

A 366 G1, SA 365 N1, AS 365 N2, AS 365 N3, EC 155 B and EC 155 B1 helicopters, all serial numbers.



This Emergency Airworthiness directive (EAD) is issued following the loss of the tail rotor pitch control on a helicopter during a landing phase. This loss of pitch control is due to significant damage to the bearing of the control rod in the Tail Gear Box (TGB).

This EAD supersedes EAD n° 2006-0051-E and introduces the following actions:

- maintaining TGB oil level at the maximum level,
- check of axial play in the tail rotor pitch control rod bearing, in case of metallic particle(s) detected at the TGB magnetic plug.

The loss of tail rotor pitch control can lead to the loss of yaw control of the helicopter.

The original issue of this EAD was withdrawn as the referenced Alert Service Bulletins had not been issued yet. This revision contains no changes to the original required actions.

Effective Date:	01 September 2006
	The following measures are rendered mandatory from the effective date of this AD:
	1. During the check after the last flight of the day (ALF-check) or during each flight-related check (15 hours or 7 days), according to the helicopter version, ensure that the TGB oil level is at the maximum level, in compliance with the instructions given in paragraph 2.B.1. of the referenced EUROCOPTER Alert Service Bulletins (ASP) corresponding to the helicopter version.
	2. For TGBs not equipped with a magnetic plug with electrical indicating:
	At the latest at the next scheduled check of the magnetic plug following the effective date of this AD and the lever 25 flying hours,
	Ensure that there are no chips at the regness plus in empliance with the instructions given in paragraph 2.B.2. If the referenced ASBs, corresponding to the helicopter version.
	2.1. If there are no chips, recume flights:
	2.2. If there are chips, take the measure to be applied after chip detection at the mannetic type, in compliance with the instructions given in paragraph 2.B.2.b) the referenced ASBs corresponding to the helicoper vision.
Compliance:	2.2.1. If the quantity of chips is above the removal criteria, replace the GB with a TGB in airworthy condition.
Compliance.	2.2.2. the quantity of chips is below the removal criteria, carry of a tactile check for absence of axial play in the Tail Roter Hub (TRH) pitch change spider, in compliance with the instructions given in paragraph 2.B.2.b.2) of the referenced ASBs, corresponding to the helicopter version.
	2.2.2.1. If there is no axial play in the TRH pitch change spider, resume flights.
3	2.2.2.2. If axial play is found in the TRH pitch change spider, replace the bearing of the pitch control rod, in compliance with the instructions given in paragraph 2.B.2.b.2) of the referenced ASBs, corresponding to the helicopter version.
	3. For TGBs equipped with a magnetic plug with electrical indicating:
	At the latest at the next scheduled check of the magnetic plug, or after illumination of the TGB "CHIP" warning light, following the effective date of this AD, and then every 100 flying hours and after each illumination of the TGB "CHIP" warning light,
	Ensure that there are no chips at the magnetic plug, in compliance with the instructions given in paragraph 2.B.2. of the referenced ASBs, corresponding to the helicopter version.
	3.1. If there are no chips, resume flights.

	3.2. If there are chips, take the measures to be applied after chip detection at the magnetic plug, in compliance with the instructions given in paragraph 2.B.2.b) of the referenced ASBs, corresponding to the helicopter version.
	3.2.1. If the quantity of chips is above the removal criteria, replace the TGB with a TGB in airworthy condition.
	3.2.2. If the quantity of chips is below the removal criteria, carry out a tactile check for absence of axial play in the Transph change spider, in compliance with the instructions give to paragraph 2.B.2.b.2) of the referenced ASBs corresponding to the helicopter version.
	3.2.2.1. If there is no axial play in the Tild pitch mange vider, resume flights.
	3.2.2.2. If axial play is found in the TRFn itch stage spider, replace the bearing of the pitch change control rod, in compliance with the instructions given in paragraph 2.B.2.b.2) of the reference LASPs, corresponding to the helicopter version.
Ref. Publications:	EUROCOPTER AS 6.5 Alert Service Falletin No. 05.00.54, or EUROCOPTER AS 6.66 Alert Sovice Bulletin No. 05.37, or EUROCOPTER EC Alert Sovice Bulletin No. 05A015, as applicable, or all approved recisions.
	If requested the peropriately substantiated the responsible EASA ranages for the related product has the authority to accept Alternative Methods of Compliance (AMOCs) for this AD.
	The safety assessment has requested not to implement the full posultation process and an immediate publication and notification.
Remarks:	3. Enquiries regarding this AD should be addressed to AD Focal Point, Certification Directorate, EASA. E-mail: ADs@easa.europa.eu .
	4. For any questions concerning the technical content of the requirements in this AD, please contact: EUROCOPTER (STDI) - Aéroport de Marseille Provence 13725 Marignane Cedex – France; Tel.: 33 (0) 4 42 85 97 97 - Fax: 33 (0) 4 42 85 99 66; E-mail: Directive.technical-support@eurocopter.com .